

MICROCOPY RESOLUTION TESTS CHART NATIONAL BURLAND OF STANDARD AND HE



NPS-72-86-001

NAVAL POSTGRADUATE SCHOOL Monterey, California



THESIS

HIGH ENERGY ELECTRON RADIATION DEGRADATION OF GALLIUM ARSENIDE SOLAR CELLS

bу

Don W. Gold

March 1986

Thesis Advisor:

A. E. Fuhs

Approved for public release; distribution unlimited

Prepared for: Department of the Navy

Commander Space and Naval Warfare

Systems Command, PDW 106-4 Washington, D.C. 20363-5100



86 6 10 116

NAVAL POSTGRADUATE SCHOOL Monterey, California

Rear Admiral R. H. Shumaker Superintendent

David Schrady Provost

This thesis was prepared in conjunction with research supported in part by Commander Space and Naval Warfare Systems Command, PDW 106-483, Washington, D.C.

Reproduction of all or part of this report is authorized.

Released as a Technical Report by:

John N. Dyer

STATE OF THE PROPERTY OF THE P

Dean of Science and Engineering

T	27		7	7 24 24 2	TION O		
٠.		# 1 ▼	1 I A		LTICAM /	3E TM16	DACE

REPORT DOCUMENTATION PAGE					
18 REPORT SECURITY CLASSIFICATION UNCLASSIFIED		16. RESTRICTIVE MARKINGS			
2a SECURITY CLASSIFICATION AUTHORITY	3 DISTRIBUTION / AVAILABILITY OF REPORT				
26 DECLASSIFICATION / DOWNGRADING SCHEDU	LE	Unlimited	l Distribut	tion	
4 PERFORMING ORGANIZATION REPORT NUMBER	R(S)	5 MONITORING	ORGANIZATION	REPORT NUMB	ER(S)
NPS-72-86-001					
6a, NAME OF PERFORMING ORGANIZATION	6b OFFICE SYMBOL (If applicable)	7a. NAME OF MI			néana Contana
Naval Postgraduate School	Code 62	Commander Space and Naval Warfare Systems Command, Code PDW 106-483			
6c. ADDRESS (City, State, and ZIP Code)	-	76 ADDRESS (Cit	-		
Monterey, CA 93943-5000		Washingto	on, D.C. 2	20363–5100	ŀ
Ba NAME OF FUNDING/SPONSORING	Bb OFFICE SYMBOL	9 PROCUREMEN	TINSTRUMENT	IDENTIFICATION	NUMBER
ORGANIZATION	(If applicable) PDW 106-483	N4175686W	JR67571		
8c ADDRESS (City, State, and ZIP Code)		10 SOURCE OF		ERS	
Washington, D.C. 20363-51	00	PROGRAM	PROJECT	TASK	WORK UNIT
		ELEMENT NO	NO	NO	ACCESSION NO
11 TITLE (Include Security Classification)			l	L	
HIGH ENERGY ELECTRON RADI	ATION DEGRADATI	ON OF GALLIU	M ARSENIDE	E SOLAR CEI	LLS
12 PERSONAL AUTHOR(S) Gold, Don W.					į
'3a TYPE OF REPORT '3b TIME C Master's Thesis FROM		14 DATE OF REPO	RT (Year, Mont	h, Day) 15 PA	IGE COUNT
16 SUPPLEMENTARY NOTATION		L —			
17 COSATI CODES	18 SUBJECT TERMS	Continue on revers	e if necessary a	ind identify by	block number)
F ELD GROUP SUB-GROUP	Gallium Arseni	de Solar Cel	ls; Solar	Cell Test:	ing
	Radiation Degr	adation; Ele	ectron Kadi	lation Degi	radation
19 ABSTRACT (Continue on reverse if necessary	and identify by block i	number)			
A need existed to perfo	rm high energy	electron irr		-	
gallium arsenide solar cell					
test facility was construc the Naval Postgraduate Sch	. •				,
to selected fluence levels					
were calculated, and it was	_	•	-	•	
by 50 % following a cumulative irradiation by electrons to a total fluence of					
$1 \times 10^{13} \text{ e/cm}^2$					
(101+1. 15+1 tow.)					
20 D S PIBUTION / AVAILABILITY OF ABSTRACT 21 ABSTRACT SECURITY CLASSIFICATION					
SAME AS A 22 NAME OF RESPONSIBLE INDIVIDUAL RESPONSIBLE INDIVIDUAL	RPT DTIC USERS	226 TELEPHONE (include Area Co	I	
A. E. Fuhs 408-646-2948 Code 72					

All other editions are obsolete

Approved for public release; distribution unlimited

High Energy Electron Radiation Degradation of Gallium Arsenide Solar Cells

by

Don William Gold Lieutenant, United States Navy B.S., United States Naval Academy, 1978

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN ELECTRICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL March 1986

Author:	week.
•	Don W. Gold
Approved by:	allen E, Fichs
pprovou by	Allen E. Fuhs, Thesis Advisor
	Ja Bowe
-	John P. Powers, Second Reader
•	Harriett B. Rigas, Chairman, Department of Electrical and Computer Engineering
_	(MONA)
	John N. Dyer
	Dean of Science and Engineering

ABSTRACT

A need existed to perform high energy electron irradiation experiments on gallium arsenide solar cells. To support this research, an automated solar cell test facility was constructed, gallium arsenide solar cells were obtained, and the Naval Postgraduate School LINAC facility was utilized to irradiate the cells to selected fluence levels at 20 MEV energies. Equivalent damage coefficients were calculated, and it was found that the average maximum power output decreased by 50 % following a cumulative irradiation by electrons to a total fluence of 1 x 10¹⁵ e/cm².

Accession For	
NTIS GEA&I	M
DTIC TIB	7
Unancia prod	
Ju. tiflestion_	
Avoil	```s
Dist (



TABLE OF CONTENTS

I.	INTE	RODUCTION	- 7
II.	SOL	AR CELL LABORATORY SYSTEM DESIGN	10
	Α.	SYSTEM STRUCTURE	10
	В.	SYSTEM OPERATION	14
	c.	COMPUTER INTEGRATION	19
III.	SOFT	TWARE DESIGN	20
	Α.	INITIAL PLANNING	20
	В.	SOLAR CELL TEST ROUTINE	20
IV.	PRE-	-IRRADIATION DISCUSSION	27
	Α.	CELL SELECTION	27
	B.	PRE-IRRADIATION CELL TESTS	29
V.	TEST	PROGRAM	30
	Α.	LINEAR ACCELERATOR DESCRIPTION	30
	В.	SOLAR CELL IRRADIATION FIXTURE	30
	С.	IRRADIATION PROCEDURES	30
VI.	DISC	CUSSION OF TEST RESULTS	35
	Α.	DAMAGE EQUIVALENT 1 MEV FLUENCE	35
	В.	ERROR ANALYSIS	37
VII.	CONC	CLUSIONS AND RECOMMENDATIONS	47
APPENDI	X A	TECHNICAL SPECIFICATIONS	49
APPENDI	X B	OPERATIONAL AND CALIBRATION PROCECURES	60
ADDENIO	TY C	SOFTWARE FLOWCHART	6.7

APPENDIX D: S	SOLAR CELL TEST PROGRAM	81
APPENDIX E: I	PRE-IRRADIATION CELL TEST RESULTS	95
APPENDIX F: I	POST-IRRADIATION CELL TEST RESULTS	104
LIST OF REFER	ENCES	131
BIBLIOGRAPHY -	***************************************	133
INITIAL DISTRI	IBUTION LIST	134

ACKNOWLEDGMENT

The author wishes to thank Distinguished Professor A. E. Fuhs for his guidance in the preparation of this thesis, and Mr. Don Snyder for his time spent operating the LINAC. The author also thanks his brother Bill for providing technical advice. Additionally, the author wishes to thank his wife and daughter for their support and humor and making the entire experience enjoyable.

SERVICE PROPERTY PROPERTY PROPERTY PROPERTY AND PROPERTY.

I. INTRODUCTION

Extensive research has been performed on gallium arsenide (AlGaAs-GaAs) solar cells concerning their superior hardness with respect to electron radiation degradation when compared with silicon solar cells. Because of known extraterrestrial electron energies, the majority of published irradiation studies have been performed using 1 million electron volts (MEV), or lower, electron energies. Further, much of the research has utilized non-production type GaAs solar cells. Thus, a need existed to perform electron irradiation experiments on production run cells at electron energies greater than 1 MEV.

To support this research, the capability to measure accurately the electrical characteristics of illuminated solar cells was required. Additionally a source of high energy electrons with which to perform irradiations was needed.

A facility that provides electrons in the 20 to 100 MEV energy spectrum is available for use through the Naval Postgraduate School's Department of Physics. Thus, the task remained to construct an operating solar cell laboratory to assess cell parameters before and after electron irradiations.

The majority of the components with which to equip such a laboratory had previously been purchased; See Mabie [Ref. 1: pp. 35-48]. This equipment included an IBM PC/XT computer, a Kratos SS 2500 Solar Simulator, an HP 7475A graphics plotter, an IBM 4 channel, 12 bit analog-to-digital converter card, and a National Instruments general purpose interface bus (GPIB) card. To complete the laboratory, a decision was made to purchase an HP 6825A/59501B bipolar power supply combination to function as an active load for the solar cells and an HP 3478A GPIB-capable digital multimeter to measure cell current.

A report of the integration of the above components into a cohesive solar cell testing system, and the subsequent purchase, irradiation, and testing of AlGaAs-GaAs solar cells is the subject of this thesis.

Chapter II discusses the system structure and computer interface.

Chapter III provides narrative concerning the software written to support the automated solar cell testing routine.

The method used to select the gallium arsenide cells for the irradiation experiments and the subsequent test program is covered in Chapters IV and V, respectively.

A discussion of the test results, including calculated equivalent damage coefficients and an analysis of the possible errors in the data is provided in Chapter VI.

Conclusions based on the results of the irradiated cell tests, as well as a recommendation for further solar cell testing, is provided in Chapter VII.

II. SOLAR CELL LABORATORY SYSTEM DESIGN

A. SYSTEM STRUCTURE

CONTRACTO CONTRACTO

The solar cell testing system consists of five main components. The first is the IBM PC/XT computer. This component, with its associated software, controls the data acquisition and power supply components. Included within the PC/XT component is the Hewlett Packard 7475A plotter. A more complete description of the IBM PC/XT computer and the HP plotter may be found in Mabie [Ref. 1: pp. 38-39].

The Data Acquisition and Control element is the next major component of the system. This component consists of a 12 bit, 4 channel analog-to-digital converter (DAC adapter), internal to the IBM PC/XT, and a Hewlett Packard model 3478A Digital Multimeter, operating over an IEEE-488 General Purpose Interface Bus (GPIB). Both devices are controlled by the PC/XT and provide voltage and current measurements of the cell under test. Technical specifications for these devices may be found in Appendix A.

The third component of the test system is the combination of the Hewlett Packard model 59501B Power Supply Programmer and the model 6825A Bipolar Power Supply. The power supply programmer operates over the IEEE-488 interface bus and allows remote programming of the 6825A (operating in

the current sink mode). Technical specifications for these two components are found in Appendix A.

The fourth component of the test system is the Kratos Model SS 2500 Solar Simulator. A complete description of the Kratos light source may be found in Appendix A.

A determination was made of the spectral output of the Kratos light source to verify close approximation to the solar constant both spectrally and in magnitude. This was accomplished by taking short circuit current readings from six different gallium arsenide solar cells illuminated by the sun at local noon and illuminated by the Kratos source. Before the readings were taken, one of six bandpass filters was placed in front of the cell to provide a wavelength dependent output. Thus, a relative spectral output of the Kratos source was determined. This procedure may be algebraically verified since the cell output current, i, is equal to the cell area, A, times the integral over the total input wavelength of the products of the cell's spectral response, \mathbf{R}_{λ} , the filter transmission response, \mathbf{T}_{λ} , and the input spectral intensity, I_{λ} from the simulator $\left(I_{\lambda}^{\text{sim}}\right)$ or from the $sun(I_{\lambda}^{sun})$. This may be written as

$$i = A \int_{\lambda_1}^{\lambda_2} R_{\lambda} T_{\lambda} I_{\lambda} d\lambda \qquad (1)$$

For the relative output current then,

$$\frac{i_{\lambda}^{sim}}{i_{\lambda}^{sun}} = \frac{A \int_{\lambda_{1}}^{\lambda_{2}} R_{\lambda} T_{\lambda} I_{\lambda}^{sim} d\lambda}{A \int_{\lambda_{1}}^{\lambda_{2}} R_{\lambda} T_{\lambda} I_{\lambda}^{sun} d\lambda}$$
(2)

The limits of integration are defined by the bandpass of the optical filters ($\Delta\lambda$). Since R_{λ} and I_{λ} do not vary significantly over the narrow band of the filters, ($2\Delta\lambda$), equation (2) may be rewritten as

$$\frac{i \sin \alpha}{\lambda} = \frac{A R_{\lambda} I_{\lambda}^{\sin \alpha} \int_{\lambda - \Delta \lambda}^{\lambda + \Delta \lambda} d\lambda}{A R_{\lambda} I_{\lambda}^{\sin \alpha} \int_{\lambda - \Delta \lambda}^{\lambda + \Delta \lambda} d\lambda} \tag{3}$$

Equation(3) may be simplified by cancelling like terms to arrive at a final result

$$\frac{i \overset{\text{sim}}{\lambda}}{\underset{\lambda}{\text{isun}}} = \frac{I_{\lambda}^{\text{sim}}}{I_{\lambda}^{\text{sun}}} \tag{4}$$

For each filter bandpass 12 short circuit current values were averaged for solar illumination and 25 short circuit current values were averaged for simulator illumination. Standard deviations for each illumination method and bandpass were then computed. The values for isim and isum were plotted and may be seen in Figure (1) as a function of wavelength. Error bars are included to indicate a 95 % confidence limit. The bandpass of each filter was nominally 10 nanometers and this is also indicated in the figure. The tabulated data may be seen in Table 2-1.

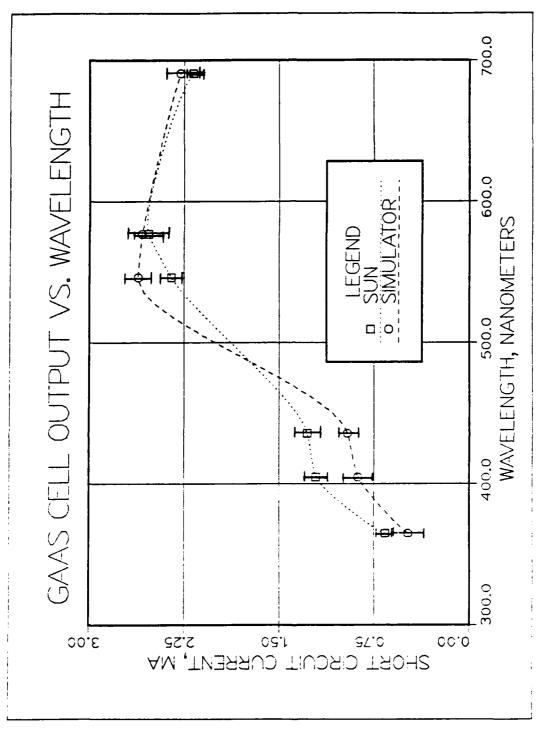


Figure 1. Gallium Arsenide Solar Cell Output (i^{sim} and i^{sun}) as a Function of Wavelength.

	TABLE 2-1					
	KRATOS SPECTRAL OUTPUT RESULTS					
	SUN SIM					
λ nM	i mA	σ	i mA	σ	;sim ;sun	
365.0	.661	.026	.480	.060	.7262	
404.7	1.21	.047	.879	.067	.7264	
435.8	1.28	.061	.963	.045	.7523	
546.1	2.35	.047	2.61	.053	1.11	
577.0	2.53	.059	2.58	.097	1.02	
690.7	2.18	.036	2.28	.062	1.04	

The fifth component of the system is the combination of the temperature control water circulator and the cell test block. Both of these devices are described in detail in Mabie [Ref. 1: p. 37]. The test block utilizes a four point electrical connection to the solar cell. That is, two separate pairs of wires are used to measure cell voltage and current. This connection effectively eliminates the problem of correcting for lead and contact resistance in the cell measurements [Ref. 2: p. 389].

The temperature control maintains a constant test block temperature to within +/- 0.5 degrees Celsius, over the range -20 to +70 degrees Celsius. The single deficiency to the circulator is that it requires approximately 1 hour to stabilize at the desired operating temperature.

B. SYSTEM OPERATION

The theory behind testing an active device such as a solar cell is discussed extensively in Rauschenbach [Ref. 2: pp. 376-406]. Basically, by varying the resistance of a load from zero to infinity, the solar cell load is changed from a short circuit to an open circuit. Thus, short circuit current (I_{SC}) and open circuit voltage (V_{OC}) may be determined by appropriate measurements. These values are indicated on the current vs. voltage (I-V) curve shown in Figure (2). As the load is varied, the curve that results quantitatively describes the characteristics of the

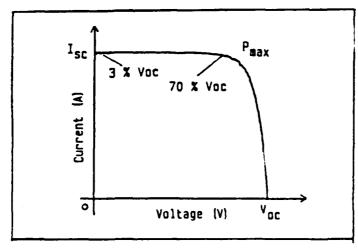


Figure 2. Typical Characteristic Curve for an Illuminated Solar Cell.

cell. Instead of using a passive load however, an active load may also be used. The advantage is that the load resistance may actually be varied between zero and infinity, which is not possible with a practical potentiometer. The

active load used in this system is the Hewlett Packard 6825A bipolar power supply. The source and sink characteristics for this device are included in the technical specifications in Appendix A.

For this application the power supply is operated in the second quadrant, functioning as a current sink.

The block diagram for the electrical components is shown in Figure (3). The test module contains switches for selection of either N/P or P/N solar cells (allowing for reversal of cell polarity), calibration or run functions, and test or setup functions. The wiring diagram for the test module is shown in Figure (4).

CONTROL CONTROL CONTROL CONTROL CONTROL

To perform a test, the voltage of the power supply is programmed to a value equal to the measured open circuit voltage of the cell, at which time the test/setup switch is placed in the test position thereby completing the test circuit. The power supply output is stepped down while sinking cell current until the voltage measured at the solar cell is less than or equal to zero. During this stepping process, current and voltage readings are taken so as to construct an I-V curve. Once the I-V data are obtained the maximum power, voltage, and current points are determined. The maximum power point is illustrated on the I-V curve of Figure (2).

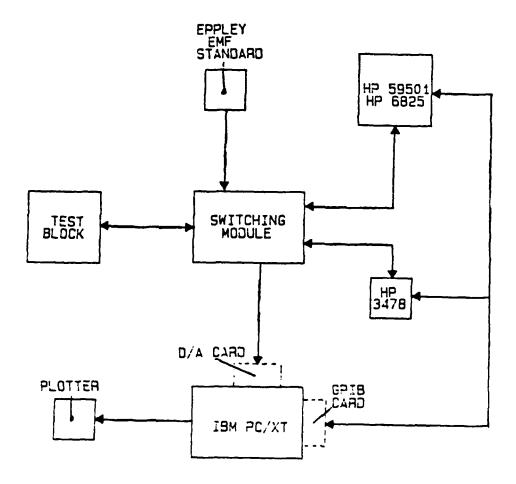


Figure 3. Block Diagram of the Electrical Components for the Solar Cell Testing System.

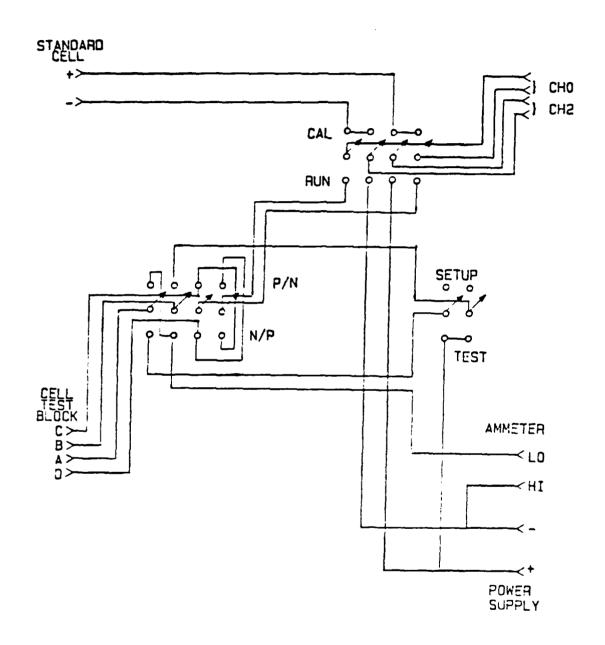


Figure 4. Wiring Diagram for the Switching Module in the Solar Cell Testing System.

DECEMBER OF THE PROPERTY OF TH

Detailed operational and calibration procedures for the system are found in Appendix B.

C. COMPUTER INTEGRATION

The inclusion of the computer into the testing system allows for repeatability, accuracy, and speed in data acquisition and reduction. In addition to the standard graphics, serial, and parallel I/O cards, the IBM PC/XT is equipped with a National Instruments General Purpose Interface Bus (GPIB) card. This device allows communication to and from the power supply programmer and the digital multimeter (acting as the data acquisition component for cell current). The other unique card in the computer is the IBM 4 channel analog-to-digital converter. This device allows data acquisition of the cell and power supply output voltages. The program used to control the computer is discussed fully in Chapter III.

III. SOFTWARE DESIGN

A. INITIAL PLANNING

THE PROPERTY OF THE PARTY OF

The prerequisites for the test program were that it must be fast yet flexible enough so that a novice at solar cell testing could obtain accurate and repeatable results. program was required to contain the separate drivers for the General Purpose Interface Bus (GPIB) and the Analog-to-Digital (A/D) cards which had previously been purchased. The novice operability requirement dictated a menu-driven program to enable easy access to the various test features These features were as follows: calibration, as desired. test, and output. The program developed into nine subroutines: two for input, one for cell testing, five for output, and one to exit to the disk operating system (DOS). Each of these nine subroutines is menu selected and returns to the menu upon completion with the exception of the exit routine. The final flow chart to support the cell test program is included as Appendix C.

B. SOLAR CELL TEST ROUTINE

Based on the above prerequisites and flowchart, the program was developed in three phases: input routines, data acquisition routine, and output routines. The line numbers in the following paragraphs refer to the program in Appendix D.

1. Input Routines

The input routines comprise two of the nine possible menu selections. The first is the data acquisition calibration routine. This selection is required prior to running a solar cell parameter test. The routine utilizes an Eppley low-temperature-coefficient standard of EMF as a known voltage source and then calculates the two offset values for channels 0 and 2 of the DAC adapter based on this known voltage. As was discussed in the introduction, the 12 bit A/D is set for a +/- 5 volt input. Therefore the voltage measured for any given input is equal to

$$V_{\text{meas}} = \text{digital word } X \text{ range } / 2^{12} - V_{\text{offset}}$$
 (5)

The digital word is a value dependent on the input voltage present at the A/D input when it is programmed to sample and varies from 0 to 4096. Range is a selectable feature and is either 10 or 20. $V_{\rm offset}$ is a value dependent on the selected A/D range. For the +/- 5 volt range used here, $V_{\rm offset}$ should be 5 volts. It was found that the offset value used with the DAC adapter calculations was not equal to 5 volts and actually varies over a period of time. This variability dictated a calibration routine resident in the test program and is found within lines 2620 - 2980. Use of this routine will be discussed in Appendix C.

The second input routine is found within lines 6320 to 6620. The floppy disk read routine allows data previously recorded on a 5 inch floppy disk to be read into the program and thus to be accessible to the output and graphing routines. This feature allows pre-irradiation and postirradiation I-V curves to be plotted on the same set of axes.

2. Data Acquisition Routine

Contractor Acceptable Contractor Contractor Contractor

The data acquisition routine involves the cell test subroutine and begins at line number 3000. Operator instructions are displayed to place the cell in the test fixture as well as to input the cell ID and size. Since the gallium arsenide solar cells are sensitive to current surges, an open circuit voltage condition is required to exist before the circuit of Figure (3) is completed, [Ref. 3]. Lines 3150 to 3260 determine $V_{\rm OC}$ generated by the solar cell, through the A/D card. Lines 3280 to 3330 then drive the output of the power supply to a value equal to the cell open circuit voltage.

Verification that the power supply output is within 1.5 millivolts (mV) of the cell $V_{\rm OC}$ is performed within lines 3340 - 3490. The 1.5 mV figure arises as a result of the inherent resolution in the 6825A power supply. For this application the best resolution possible is 1.5 mV which becomes the smallest voltage deviation achievable. If the

difference between the $V_{\rm OC}$ and the power supply output is greater than 1.5 mV, the program loops and again samples $V_{\rm OC}$, drives the power supply to the appropriate voltage, and then tests the difference. If this loop occurs 5 times then an external problem exists and the program returns control to DOS. The external conditions that might cause this condition are covered in Appendix B.

The next portion of the program prompts the user to close the test/setup switch and complete the test circuit. At this time the circuit current should be nearly zero and is measured in lines 3630 - 3650. If the current is negative, the power supply is supplying current and the solar cell is sinking power (an undesirable condition). The power supply output is stepped 1.5 mV lower, and the current is again sampled to verify a zero or small positive value. If this condition still does not exist, then an additional 1.5 mV is subtracted. This process continues until a zero or small (nominally 0.5 milliamps) positive current exists, and the program then prompts the user to close the test/setup switch and complete the test circuit.

The next portion of the cell test subroutine is the data acquisition routine. It is composed of 4 separate modules, three of which contain instructions to measure cell voltage and current, calculate the cell's power at the data

point, and downstep the power supply. The fourth module allows for a transient-free disconnect.

The first module (lines 3730 - 4020) measures the cell parameters from $V_{\rm OC}$ to 70% of $V_{\rm OC}$. As shown in Figure (2), this interval includes the maximum power point for the cell. Since the slope of the curve is steep in this region, the power supply steps in 3 mV increments.

ではないのからの 人名の名はない いんじんかいかん

The second module (lines 4030-4260) is identical to the first except that it operates between the 70 % and the 3 % $V_{\rm OC}$ points. The curve is relatively flat here and thus the program steps in 23 millivolt steps to reduce test time.

Each of the first two modules contain a delay loop (lines 3970 - 4000 and 4210 - 4240) to allow a settling time (approximately 40 milliseconds) between measurements.

The third module (lines 4270 - 4470) operates from the 3 percent point to the point at which the measured cell voltage is less than or equal to zero, that is, the cell short circuit current value. This module also steps in 3 mV increments.

The fourth module (lines 4490 - 4600) drives the power supply back to an open circuit voltage condition; $V_{\rm OC}$ is required for a transient free disconnect. The test circuit is opened by the operator, and the program returns the power supply to a zero volts output condition (this is a turn-on/turn-off requirement for the power

supply). The remainder of the fourth module (lines 4730-4780) determines the maximum power, maximum voltage, and maximum current point. Lines 4790 - 4880 calculate 3 significant digits for the cell parameters.

3. Output Routines

The output portion of the program comprises five subroutines. The first, lines 4910 - 5010, prints the voltage, current, and power at each data point measured onto the color monitor, or, by using the control-print screen keys, the data may be printed on the Epson printer for permanent copy.

The second subroutine, lines 5030 - 5150, plots the solar cell I-V curve on the color monitor. This provides a fast, graphic presentation of the cell characteristics.

The third subroutine is the plotting routine for use with the Hewlett Packard 7475A plotter and is found within lines 5170 to 5970. This routine draws a border, axes, labels, and plots the characteristic I-V curve. The numerical parameters for the cell as well as identifying data are included as well. Several questions are posed to the operator during this routine to allow for multiple plots on a common set of axes. This allows an I-V curve for an irradiated cell to be drawn on the same axes as an I-V curve for the same un-irradiated cell. This feature requires the use of the floppy disk read subroutine as discussed above.

The fourth output subroutine writes the cell voltage, current, and power at each data point measured to a floppy disk. This allows a permanent record of cell data to be maintained. The subroutine also includes writing the cell ID and size to the floppy disk. This routine is found within lines 5990 - 6120.

The last output subroutine prints the calculated parameters of +' solar cell on the CRT display. This allows a rapid determination of the cell $V_{\rm OC}$, $I_{\rm SC}$, maximum power, voltage and current point, as well as the cell fill factor and efficiency. This print may be inspected in lines 6140 - 6300.

IV. PRE-IRRADIATION DISCUSSION

A. CELL SELECTION

Production run gallium arsenide solar cells were desired as test specimens for the irradiation experiments. As opposed to a limited production run cell (i.e., those specifically designed and manufactured for experimental use only), irradiation of production run cells could be expected to provide more general results for a specific design. The difficulty found in this approach was that none of the major manufacturers contacted were producing gallium arsenide cells in production quantities at the time cell selection was made.

The Applied Solar Energy Corporation (ASEC), was apparently the closest to producing a "production" cell and thus 20 gallium arsenide cells were ordered. ASEC supplied complete I-V characteristics for each cell as a means of comparison. These 20 cells were specified to have an efficiency of 16% or greater. ASEC also supplied 5 "control" cells, that had a stated efficiency between 15 and 16 per cent. Table 4-1 provides complete technical data for these 2 by 2 centimeter solar cells. Figure (5) is a cross sectional view for this particular cell.

TABLE	4-1			
ASEC P/N 2 x 2 centim	neter GaAs SOLAR CELLS			
Substrate thickness 300 uM N layer thickness 9 uM P junction layer thickness-Zn doped 0.5 uM P window - $Al_xGa_{1-x}As$ (x ~.869) thickness 0.1 uM P contact material Au - Zn - Ag N contact material Au - Ni - Ge - Ag Anti-reflection coating TiO_2 , Al_2O_3				
[From Reference (4)]				

COMPANY CONTRACTOR OF CONTRACT CONTRACT CONTRACTOR CONT

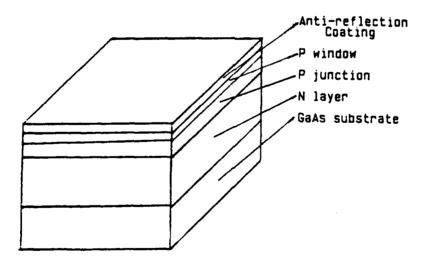


Figure 5. Cross-Sectional View for the ASEC AlGaAs-GaAs Used in this Report.

B. PRE-IRRADIATION CELL TESTS

Current-voltage characteristics for each solar cell were generated, recorded on floppy disks, plotted, and then compared with the ASEC data. The results were within 1 % of the supplied data and are tabulated in Table 4-2. Appendix D contains the graphical I-V characteristics prior to irradiation for each cell tested. The temperature for all tests was 28 degrees (+/- 0.5 degrees) Celsius.

TABLE 4-2 CELL CHARACTERISTICS (PRE-IRRADIATION) (measurements made @ 28 degrees Celsius) Cell ID Eff Voc Isc Pma x Vma x Imax FF mVmΑ mW mV mΑ ξ ASEC 2 985 117 89 815 109 .772 16.5 993 ASEC 3 115 89 817 110.782 16.6 ASEC 4 990 116 108 88 816 .767 16.3 ASEC 5 963 126 90 797 113 .776 16.6 983 117 89 825 108 .773 ASEC 6 16.4982 118 ASEC 11089 808 .767 16.5116 111 ASEC 8 975 89 806 .786 16.5 ASEC 9 948 119 89 799 $\overline{111}$.787 16.4 ASEC 10 957 116 89 ŔΙÚ 110 .803 16.5

V. TEST PROGRAM

A. LINEAR ACCELERATOR DESCRIPTION

The Naval Postgraduate School Linear Accelerator (LINAC) is a traveling wave type accelerator useful for experimenting with electron energies ranging from 20 to 100 million electron volts (MEV). The average current in the electron beam is 0.1 microamperes and has a pulse frequency of 60 Hertz. More detailed descriptions of the linear accelerator may be found in the bibliography.

B. SOLAR CELL IRRADIATION FIXTURE

Figure (6) is a photograph of the test fixture constructed for this experiment with the three test stations filled. The aluminum fixture supports a phenolic rectangle which in turn holds 3 pairs of cell retaining blocks. The retaining block system was designed to allow accurate, repeatable placement of the solar cells as well as minimizing the possibility of cell breakage.

Between the first and second test stations is an area devoted to a phosphorus target. This target provides a visual means of focusing the electron beam onto the cell area.

C. IRRADIATION PROCEDURES

The solar cells to be irradiated were placed in the test fixture which was then stationed in the LINAC vacuum

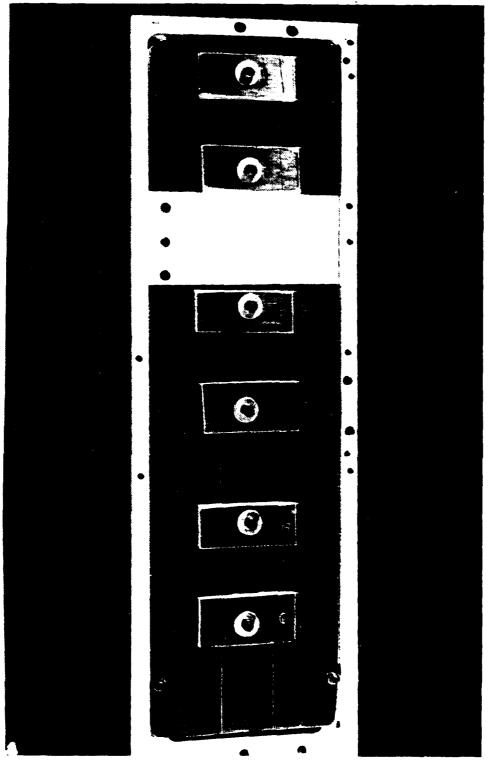


Figure 6. Photograph of the Aluminum Irradiation Test Fixture, Showing 3 Cells in the Test Stations and the Phosphorus Target.

chamber, Figure (7). A vacuum of 6.5×10^{-8} atm was established and an electron energy level was selected. The LINAC was then operated for the time required to achieve

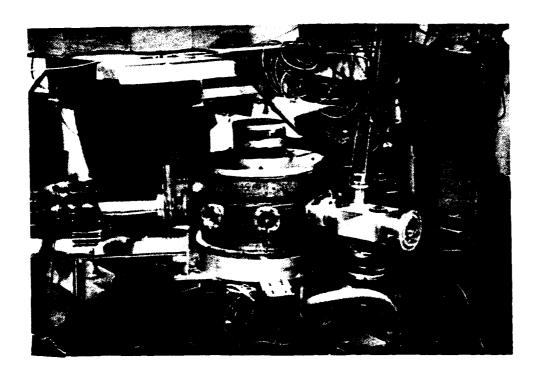


Figure 7. Photograph of the LINAC Target Chamber (center)

the desired electron fluence. The electron fluences used ranged from 1 x 10^{12} electrons/cm² to 1 x 10^{15} electrons/cm² at increasing values of 10^1 e/cm² for each successive test. Each cell was irradiated at an electron energy of 20 MEV. This energy was chosen since data are not currently available for degradation at this level; further, 20 MEV is the lower effective limit of the LINAC energy spectrum.

The fluence that a cell receives during a test is determined by the time that the electron beam is turned on. This time is measured by monitoring the charge deposited on a capacitor. Using the relationship

then the voltage across the capacitor may be calculated by

$$V = Q/C \tag{7}$$

Since the total charge deposited on the capacitor equals the fluence, ϕ (e/cm²), times the electron charge, q_e (coul/e), times the cell area, A (cm²), then a unique voltage may be calculated for given values of ϕ and C. Algebraically, this equation may be written as

$$V = \frac{\phi \ q_e \ A \ \eta}{C} \tag{8}$$

where the terms have been previously defined. The quantity η is the efficiency of the secondary emissions monitor (SEM) that measures the electron beam current and is numerically equal to 2.6%. Figure (8) shows the configuration of the LINAC target chamber during the solar cell irradiations. All irradiations were performed at normal

incidence to the cell. Characteristic curves were measured following each irradiation.

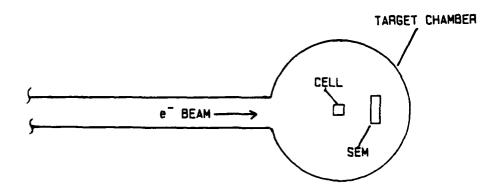


Figure 8. Configuration of Interior of Target Chamber During Solar Cell Irradiation.

Contraction (Section Contraction Property In

VI. <u>DISCUSSION OF TEST RESULTS</u>

A. DAMAGE EOUIVALENT 1 MEV FLUENCE

Table 6-1 lists the calculated mean and standard deviation values for post irradiation cell open circuit voltage, short circuit current, and maximum power as a function of 20 MEV electron fluence.

The tabulated data for the above parameters was next plotted as a function of fluence and may be seen in Figures (9) through (14), at the end of this chapter. Vertical error bars are included to indicate a 2 , 95 % confidence level. The error associated with the fluence value is less than 2 % and is not included on this scale plot.

The 1 MEV equivalent electron fluence concept was then applied to the test results. The fluence values corresponding to a 20 % degradation in $V_{\rm OC}$, $I_{\rm SC}$, and $P_{\rm max}$ were determined [Ref. 5: pp. 3.21-3.24;Ref. 6: pp. 5-16]. Next, fluence values corresponding to a 20 % degradation due to 1 Mev energy electron irradiation were extracted from Figures 3.100, 3.102, and 3.103 of Reference (5), for $V_{\rm OC}$, $I_{\rm SC}$, and $P_{\rm max}$. The fluence values extracted were then divided by the 20 MEV, 20 % degradation fluences to arrive at an equivalent damage coefficient for 1 MEV energy electrons. These values are shown in Table 6-2.

TABLE 6-1

CALCULATED MEAN VALUES FOR POST IRRADIATION CELL TESTS

(measurements made at 28 degrees Celsius)
(standard deviations in parenthesis below each parameter)

COSCIE ROZZEKKY WASHAM BESKINDO

Fluence (e/cm²)	Mean Mean Voc Isc (mV) (mA)		Mean P _{max} (mW)		
0	972.3	117.5	88.9		
	(17.2)	(1.9)	(.57)		
10 ¹²	964.3	116.4	87		
	(14.9)	(1.3)	(1.0)		
10 ¹³	944.6	113.5	82.7		
	(12.3)	(1.9)	(2.5)		
10 ¹⁴	898.3	106.6	73.6		
	(7.8)	(3.0)	(2.7)		
10 ¹⁵	775.3	76.6	44.3		
	(16.5)	(.57)	(2.1)		

TABLE 6-2							
EQUIVALENT DAMAGE COEFFICIENTS							
Parameter	∮(1) / ∮(20)						
V _{oc}	20						
I _{sc}	8						
P _{max}	3						

Thus, if it is desired to determine the fluence of 1 MEV electron radiation required to degrade a parameter to an equivalent degree of 20 MEV radiation, then the 1 MEV fluence need only be multiplied by the factor from the table.

B. ERROR ANALYSIS

The analysis of errors will be discussed in four areas: damage equivalent fluence, Kratos spectral quality, LINAC errors, and data acquisition and computation errors.

1. Damage Equivalent Fluence.

If the degradation curves for the 1 MEV [Ref. 5: pp. 3.141-3.144] and the 20 MEV cases are plotted on the same set of axes, it will be noted that they are not parallel, and therefore the concept of equivalent fluence does not strictly apply [Ref. 6: pp. 5-6]. Thus, the equivalent damage coefficient calculated for a 20 % degradation will not equal the coefficient for a 15 or 25 % degradation. The difference, however is small (< 10 %), and may be neglected given the uncertainties associated with outer space electron fluences. Of more important concern is that the gallium arsenide cell used in the figures of Reference (5), is not the same as those used in this thesis. Therefore, further study is required to determine the accuracy of these equivalent damage coefficients for this cell type vs. 1 MEV electrons.

2. Kratos Spectral Quality

A discussion was included in Chapter II concerning the spectral output of the Kratos source. Based on the values of isim/isun for the six spectral bands measured, it may be concluded that in a relative sense, the Kratos source is a fair approximation to a solar constant. The fact that the constant is not the same for the shorter and longer wavelength bands indicates that the Kratos source does not provide an equal relative spectral irradiance over those two bands. That this difference occurs does not appear to contribute a significant amount of error, however, since the pre-irradiation cell parameters differ from the supplied ASEC values by an average of 1%. Thus, the spectral quality of the Kratos source may be considered good.

3. LINAC Errors

The energy in the electron beam is determined by the voltage required by the beam deflection magnets to "steer" the beam onto the target. There is a linear relationship between the deflection magnet voltage and the energy of the electron beam, and it is this linearity that is used to establish an energy of 20 MEV. The error associated with determination of the beam energy using this procedure is less than 2 %.

As discussed in paragraph (A) above, the delivered fluence values for each irradiation could differ from the expected values by less than 2 %.

4. Data Acquisition and Computation Errors

From an examination of any one of the I-V curves in Appendices E or F, it will be noted that the curves are not smooth. This is due to several factors: (a) the inherent resolution of the power supply (1.5 mV); (b) the inherent resolution of the A/D converter (2.5 mV); (c) the averaging of five samples of cell voltage; (d) the 500 Hz sampling frequency; (e) the variations that occur in the lamp power supply causing variations in lamp intensity; and (f) the method of plotting the data.

The resolutions for the power supply and the A/D converter are essentially fixed quantities for the purposes of this solar cell testing system. Although the respective resolutions could be decreased in order to increase the range of the system, they can not be increased beyond their present values using existing components.

The number of samples averaged and the sampling frequency were both determined by extensive testing. Since much of the noise in the system occurs at 60 Hz, the sampling frequency should be greater than 120 Hz. At this frequency however, there was still an unacceptable level of noise present in the data, most easily seen in the graphical presentation. Further, the lower the sampling frequency, or the greater the number of samples averaged, then the greater is the time required to run a parameter test. Thus, a 500 Hz sampling rate and 5 samples were decided upon.

The variations that occur in the lamp intensity are related to the variations in the lamp power supply input voltage. Since the feeder to the 208 volt input circuit is shared by other devices in the building (air compressors, etc.), each time a high current device starts, there is a lowering of the input supply voltage. The deviations seen in the lamp intensity are small, averaging less than 3 % of the full scale value. There may be gross deviations however, which may require running the cell parameter test a second time.

CONTRACT CASSAGE COCCURS

The raw digital values for each data point are input to the HP 7475A plotter to arrive at a final plot. The discrete voltages that the power supply steps through are thus visible in the plots. The step size could be decreased which would minimize the unevenness in the plots, however this would have an adverse effect on the time required to run a test. One alternative would be to input analog values to a plotter which would alleviate some of the unevenness.

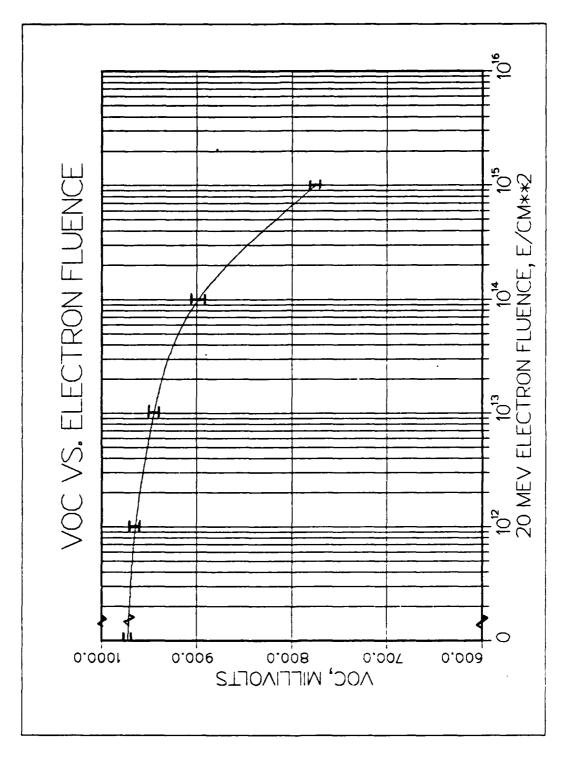


Figure 9. Solar Cell Open Circuit Voltage as a Function of 20 MEV Electron Fluence.

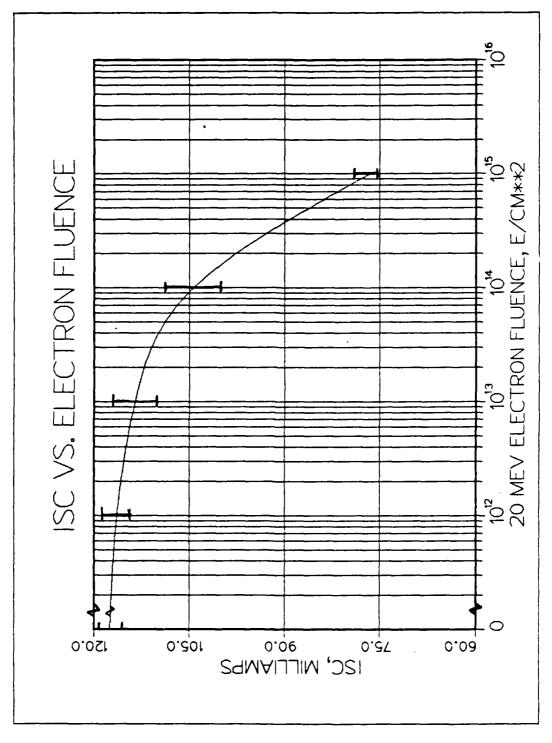


Figure 10. Solar Cell Short Circuit Current as a Function of 20 MEV Electron Fluence.

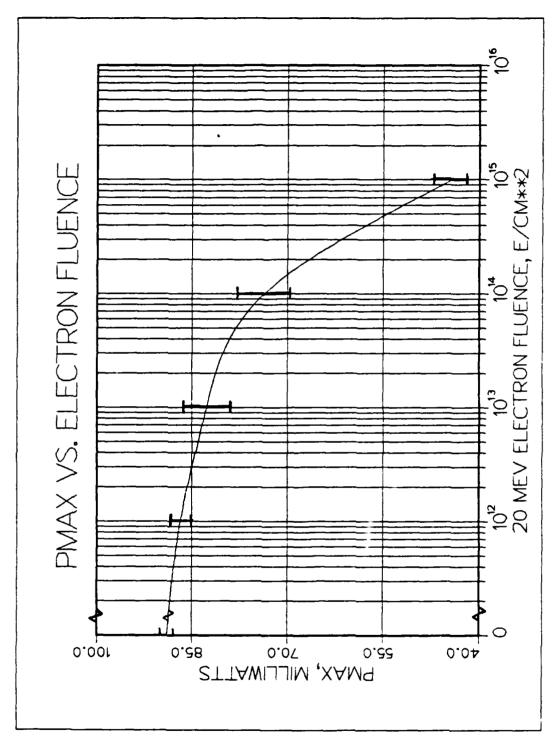


Figure 11. Solar Cell Maximum Power as a Function of 20 MEV Electron Fluence.

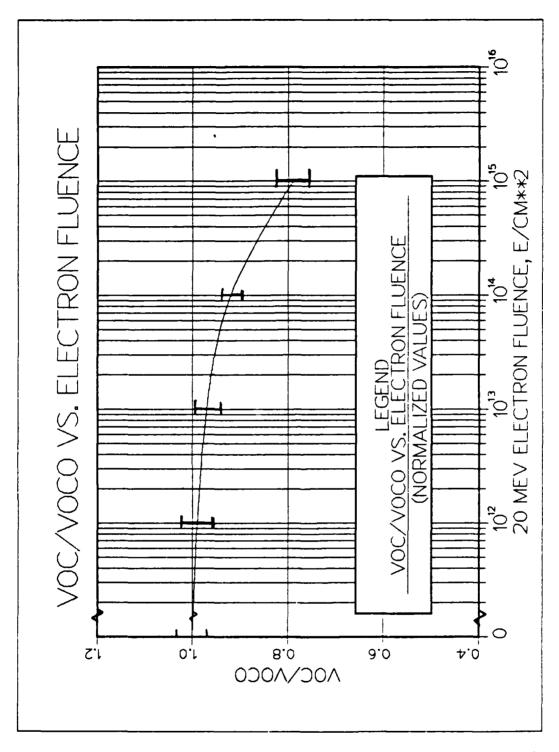


Figure 12. Normalized Open Circuit Voltage as a Function of 20 MEV Electron Fluence.

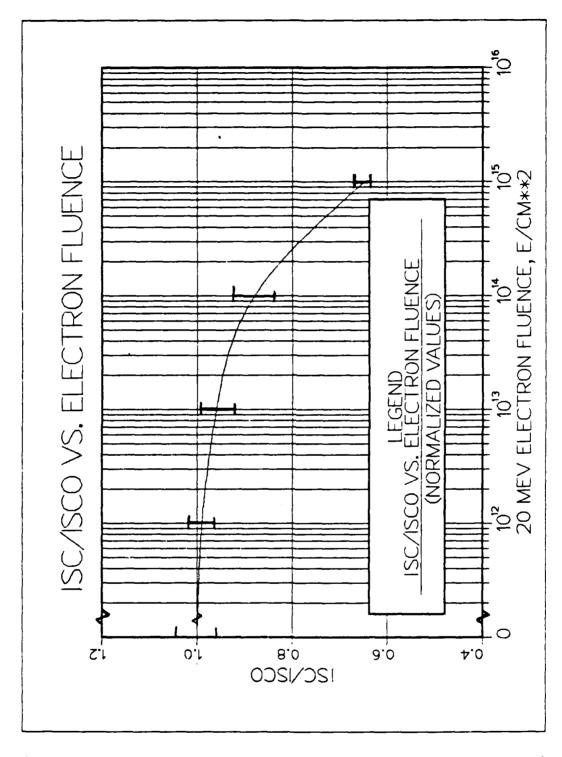


Figure 13. Normalized Short Circuit Current as a Function of 20 MEV Electron Fluence.

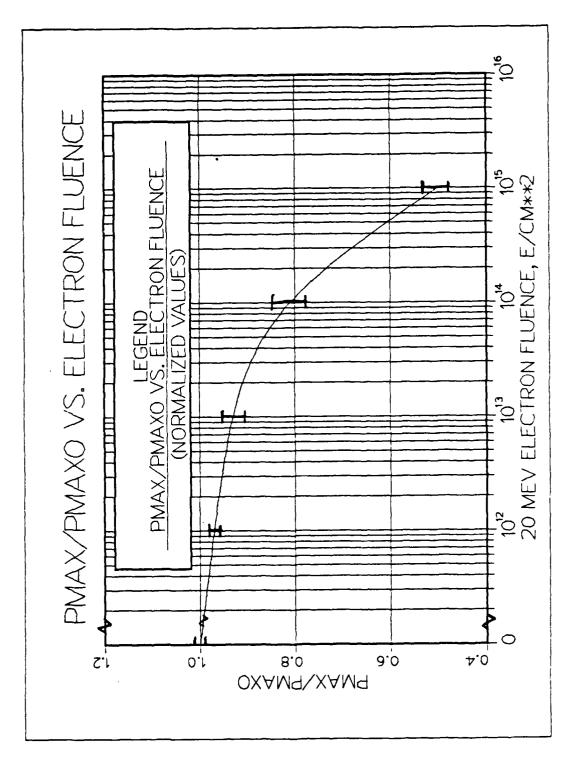


Figure 14. Normalized Maximum Power as a Function of 20 MEV Electron Fluence.

THE PROPERTY OF THE PROPERTY O

VII. CONCLUSIONS AND RECOMMENDATIONS

A. CONCLUSIONS

1. Solar Cell Testing System

Based on the excellent correlation between the ASEC measured parameters for the 25 gallium arsenide cells and the results obtained using the automated solar cell test routine, it is apparent that the solar cell laboratory provides an accurate, repeatable means of determining solar cell parameters.

2. Radiation Damage

It was found that the average maximum power output decreased by 50 % following a cumulative irradiation by electrons to a total fluence of 1 x 10^{15} e/cm².

The results of the post-irradiation testing (Table 6-1) and the equivalent damage coefficients calculated (Table 6-2), appear reasonable based on published data for lower electron energy degradation tests. Depending on the results of 1 MEV electron irradiation for this particular ASEC cell, the equivalent damage coefficients may require revision either up or down. Until this is accomplished and data published, a conservative approach should be taken when using these equivalent damage coefficients.

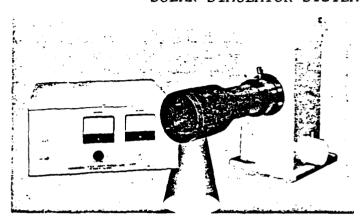
B. RECOMMENDATIONS

It is recommended that data be obtained for 1 MEV electron radiation degradation to these particular ASEC solar cells. This would enable the computation of more accurate equivalent damage coefficients.

APPENDIX A

TECHNICAL SPECIFICATIONS

TECHNICAL SPECIFICATIONS FOR THE KRATOS SS 2500 SOLAR SIMULATOR SYSTEM



The ideal Solar Simulator for any number of applications. Rain or shine.

Our Solar Simulators have found widespread acceptance—in a variety of applications—owing to their dependability, accuracy, convenience and economy. Here are just a few areas of use:

- Cosmetics and Dermatology tervithema studies, sunscreen testing, etc.;
- Solar Energy (solar cell development, solar collectors, etc.)
- Biodegradation/Solar Exposure Durability Testing
- Artificial Aging and Weathering
- Aerospace
- Medicine and Dentistry
- Environmental Simulation Champers

Five models up to 7000W.

From the relatively compact 150W system, up to the powerful 7000W system, with a 1000W, 2500W and 1000W ellipsoidal system in-between, we have the Solar Simulator that is right for your application.

Built for performance; Priced for value.

Our Solar Simulators have been designed based on existing illumination system technology—the standard for excellence. The result is a product of proven performance, at a cost within the reach of most lab budgets.

System components—lamphousing, optics, and lamp power supply—are manufactured under rigorous quality standards, ensuring optimum long term performance, Routinely.

Reproducible illumination conditions every time you turn it on.

The Xenon Arc Lamp is known for its color constancy throughout its operating lifetime. So the tests done with our Solar Simulator—even if months apart—will be done under identical illumination (try that with the sun itself!). The result is better control over your testing, and ultimately, more valid results.

Controlled spectral output.

Carefully selected optics and filters produce refined spectra closely following those of the accepted Air Mass O (AMO), AM1, and AM2 standards for solar radiation outside the atmosphere (AMO) and at ground level (AM1, AM2).

Reduced heat output.

The spectral filtering employed in all three Solar Simulators effectively attenuates the intense infrared output produced by the unfiltered Xenon arc, without distorting the desired Air Mass values. This minimizes problems associated with IR burns in cosmetic and dermatological studies, and excessive heat buildup during solar cell testing, aerospace studies, or other investigations.

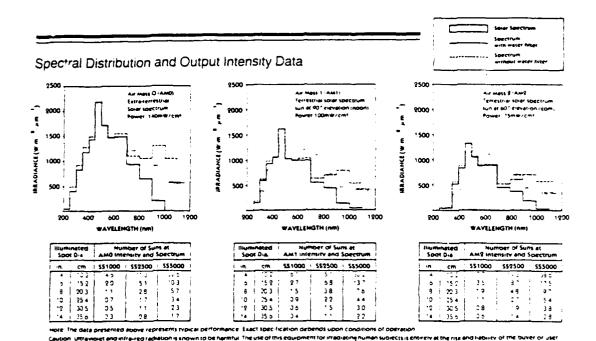
In applications where additional IR heat filtering is desired, such as high intensity illumination in cosmetic or dermatological studies, a special dichroic mirror may be used in the beam deflection tube to produce an IR-free output.

Intensity equivalent to several, even many, suns. Or, one sun at a time.

By moving the illuminated subject closer to the source, you can effectively "concentrate" the unit's power over a smaller area ... with an intensity equivalent to that of more than 50 suns! For applications such as sunscreen testing and artificial aging, this capability greatly reduces necessary exposure times,

For other applications, such as solar cell testing, where hyper-illumination is impractical, the \$\$1000 provides one solar constant at \$400 over a \$ inch diameter circle; \$\$2500 over a \$12 inch circle; \$\$55000 over a \$16 inch circle

Figure 15. Technical Specifications for the Kratos SS 2500 Solar Simulator System. [Ref. 7: pp. 34-35]



Reliable, Convenient; Easily operated

Obviously more reliable and convenient than the sun itself the Xenon Arc Lamp and Power Supply System is among the most dependable and stable widres available. Routine operation of a binar Simulator is extremely straightforward, and apart from simple amo replacement only about every 1000 mouns of operations and 5 mulator is virtually maintenance free

A modular system that changes with changing needs.

Dur Shiar Simulators provide the neximility in modular design. As needs pragret the system can be phanted without requiring investment in a new system.

The \$51,000 can ne used with any reporting 150 to 1000th this small scale doptications less powerful lamps prisode a certain degree or economy in replacement.

The \$52500 and \$\$5000 can be used with any lamp from 1000 to 7000W requiring link minor unlarger in amplifying and system colling unlarger in

Standard Solar Simulator

Our Solar Simulators include all needed components for operation. Xenon arc lamp lamp heat sinks inotified in SS 10001, amp sockets, lamp power supply, lamp housing, all potics fitter holder complete set of air mass fitters 90° heam deflection matrice and flexibose for 220ne removal.

55 1000 - 1000W Solar Simulator

Complete as described above includes cm 151 Lump militaria and 125.255 mR Lump Rower Supply. Lumps from 250 to 1000W missing used. Water fixer for IR attenuation continua.

SS 2500 – 2500W Solar Simulator Complete as deletined above includes

Complete as described above includes CH 172 Lamp Housing and LPS 400 Lamp. Power Supply Lamps from Histogram Machine used with addition of certain accessories. Includes water rifer.

55 5000 = 3000W Solar Simulator

Complete as described above includes LH 152 Lamp Housing LPS 400 Lamp Power Supply and 2 abstract rank for lamp doming Lamps from 900 b500M may be used includes water filter.

The design of this amp nousings and power supplies of their Solutions and

lator Systems permits the use of a variety or different lamps, as indicated above.

Additionally, for very small scale applications, a 150W Solar Simulator is also available.

Accessories

As indicated above leach Solar Simulation System is completely read, for operation at unsignification with a simulation with a simulation of the simulation with a simulation of applications in order accessories are reducted.

However in Certain upplications of may no Desirable to mind for the space trum in Wilme Way or Uniose trom among several options.

For example, a discription mirror may neclused in the 30° near deflection time, to reflect the 30° and discretization while dissipating the intrared paint a mirror siderial when must be sure in the 30° variety when highly discretization in the 30° variety with the 30° variety and then and discretization transmission spectral profits business are available consult us with sour approximation.

Addit hass greatring in otters sare in it has prophraems sowere beam masks and ligher accessives are assiassituble.

Figure 15. Technical Specifications for the Kratos SS 2500 Solar Simulator System. (continued)

TECHNICAL SPECIFICATIONS FOR THE HEWLETT-PACKARD MODEL 3478A MULTIMETER

	00 1	VOLTAGE			Note America						
Aput Characti	MISHES:				in dB. wi				LO IEBO.	AC reie	ection for 50
	Mezimum	1						AC !	AC	oc	
Rooge	Reading (5% Digit)	5% Digit	Resolution 4% Digit	1 3% Diget)-39-3Y	NAR	ECMR	CMR	_
30mV	= 30 3099mv	10000	1	10,47			, aig:15		150	140	
300mV	= 303 099mV	1#V	10.V	100∨			agits	59	130		
30 V	= 3 03099 V = 30 3099 V	100 V									
300 V	±303 099 V	Imv	10mV	100mV	Maximum Res	ding R	ates: +re	adings/se	C!		
ingut Resister	ee:				First readi	ng is d	correct v	when trigg	ered con	ncident v	with step inou
30mV. 30	00mV.3V ranges V ranges 10MΩ	> 10 ¹⁰ 0			The readi deing use		es are c	sependeni	on the	speed of	the controlle
Maximum Ingi	et Voltage: (non-des	tructive			Lii Fr es i		Auto Zera	i 3's Oigi	Resolut ts + 4 ;		5 - Digits
	303V rms or 450 a Earth Ground :		k		*==		: 0#	71		33 ;	4 4 2 3
Measurement	Accuracy:				*50	r z	! On	53		30	37
±:ªe of r Auto-zero	eading - number ON	of counts)			50	HZ	On	50		17	19
5% Digit Me	ide:					A	C VOLTAC	Ettrue rm	s respon	nding:	
1	Cal. Temp ±	Cal. Ter	ng. :5°C		Input Characti	~					
Range	24 Hours	96 Day	1 Yes	н							
30mv - 0	027 - 35 03	030 - 41	0 040	- 41	_		Maximum Reading			Pesolution	
3 7 9	005 - 4 00	006 + 2	; 2019 ·	- 2	Range		5 7 Digit	5	, Digit	4'1 3191	t 3's Digit
30 V I 0 300 V I 0	005 - 3 : 0 : 0055 - 2 : 0 :	007 - 2	0 020 -	- 3	300mv 3 V	3	03 099n 03099	V 1	120.0	1000	/ ·
4'n and 3'n	Digit Meda:				300 🤄				· ~ .	10m	
Accura	cy is the same as 5	a digir mo	de for 35 of	reading use	Ingut impedar	ce.					
	t for number of a se 4 counts	ounts on a	ii ranges si	10501 301114	1M0 ± 1	20 SP	unted a	v < 50pF			
The Ca	Temo Calibratio	or Tempera	rure sitne	temperature	Masimum Inc	ut Voss	age no	n-gestruc	• • •		
31 '50	environment where	e Me 3478	- was :a	orates Ca-				450. 24			
environ	ment between 20	C and 30°	: 5		41 Jr 23	o Ear	נה שנים	na :50	Cirbea•	•	
luto Zero Off					Measurement	Accura	ICY				
117 :500	for a stable environg to accuracy is 000m/y and 00y	pecification	ini 3⊝mili	. 30de	Auto zeri Auto zeri Auto - T	i jr. nursi:	5.34 on.	mber 31 - 1 7 50-+• 1 7 - • • • 1 1 2	ounts wadured scale	rv ssee	cied for sin
emaarature C	oelficient:							100m¥		nees	300v
2.0.6		ON:			Frequen	-		100mA	- 14	30¥	1007
	spiav auto zero eading – number		, ¢		10Hz 50H			4 - 163		195	1 18 - 12 2 5 1 - 1
	Range Te	mperature Co	efficient		7 (am) 5	Çver Jver	2.2	7 - 163 9 - 163 6 - 141 4 - 882	7.4	- '-:	1.13 - 1
					5/36 Hz				1.04		

Figure 16. Technical Specifications for the Hewlett-Packard Model 3478A Multimeter. [Ref. 8: pp. 1.2-1.4]

Auto Zaro C	I II.			2 Wire Ohm	ns Accuracy:						
	its) for a stable er its to accuracy s		1°C) for < 24 hrs., add for all ranges	Same as 4- wire ohms, except add a maximum of 200mth of set. On the 3M ohm Range, add, 0016% of reading and the 30M ohm Range, add, 0083%							
Temperature	Coefficient:					0083-9					
0°C :0	55°C			Auto-Zero Off:							
5% digit display, auto-zero ON For frequencies < 20kHz, ± 0.016% of reading ± 10 counts) °C					i) for a stable enviro						
For frequ	uencies > 20kHz.		reading = 10 counts) °C reading = 10 counts) °C	for 300!	nts to accuracy sper ! 3 counts for 3K!! range, and 33 coul	through 30	OK 2 range	1 300 5 8 cou			
Crest Factor	tull scale.			Temperature (113 TOP 30141.	i ange				
	de Rejection:				0°C to 55°C						
	Ω imbalance in L	o iead. > 70df	B at 60Hz		display, auto-zero reading = number		c				
Mazimum Ai	rading Rutus: Fread	ings seci			Range '	Temperature Co	aefficient				
correct seconds	range, triggered for each range o	coincident will change	s of final value, when on the step input, Add 0.6 using fast trigger (T5)	•	30() 3000 3k - 300kil	0 003 -	5 35				
Using No	ormai Trigger :T	-2 T3.	damy rest trigger (13:		30MΩ :	0 0021 - 0 021 -	05 05				
For 50	or 60Hz operati	on auto-zero	ON or OFF		,	0 02	••				
3 ; or 5 ; de	4 a digits: 1.4			Current Throu	-						
2 1 04	ynts : 0				Range	Curre	Rt .				
	RESISTANCE	2-wire () 4-w	vire 12:		30 onm	1 1 0					
gut Charact	eristics:				300 onm	1 m	Δ				
	Maximum				3K anm 30K anm						
	Reading		Resolution		300K onm	! '0.	د .				
Range	(5): Digit)		4% Digit 1 3% Digit		3M onm	1.	A				
30 n	30 3099 1	100 40	1mg 10mg		30M onm	. con	~				
	303 099 12	i , wij ,	10mil 100mil	Maximum Opi	en Circuit Valtage:						
3 ≢Ω	3 03099 kΩ	.0w/;	100mii 1 2	5 5 V							
30 kΩ 300 kΩ	30 3099 k0 303 099 k0	100mt!	10 2 100 2	Man *	0						
3MΩ	3 03099M!!	• • • •	100 0	Maximum Read	sing Kates:						
30MΩ	30 3099M0	1 100 0	1 kg - 10 kg		c voits except the						
nout Pro	tection non-des	tructive		range ad:	d 30ms (or 30Mt)	rance add i	cums ser	e3C-1:			
	ce 10 La source		•		DC CUA	RENT					
~· 5805	e to up sense :	350 - peak		input Characte	eristics						
mi or a	o to Earth Groun	; #530√ pe	ak.		Masimum						
Pasurement	Accuracy:			_	Resoing		solution				
2116.255	eading - numbe	r of counts		Range	5 > Digiti	5 » Digit	4 a Digit	3 / Digit			
Auto zero	ON E-wire one	` \$		300mA	: 203 29ama						
	i-NPUT LS mpe	pance s 3 3 :	of to scale	33044	= 3 0308A Y	13.4	1111	~			
5 : Digit Me				Maximum indi	ut: non destructive						
{ Cal. Temp : 1°C Cal. Temp. :5°C Range 24 Hours 90 Day 1 Year				3A from	< 150V source rus	ie protected					
	(4 10013		1981	Measurement	Accuracy						
711	0.023 - 1		41 1014 - 41		reading - number						
00Ω Fr-3 00 €Ω	0.0035 -		- 18 1947 - 18 - 2 2248 - 1		o oranga sa						
***	0.0052 +	3511				Cal Temp.					
OMO	1 0 036 -	2 3 366 -			Rance	90 Davs	' Year				
								_			
		Note		,	100m2		. 15 -				
- 30	M onm Range		300/0ximately		A Monte o	11 • 40 1 1 14 • 1 1 • 12) 15 - 49) 17 - 19				

COURT TO SERVICE OF THE PROPERTY OF THE PROPER

Figure 16. Technical Specifications for the Hewlett-Packard Model 3478A Multimeter (continued)

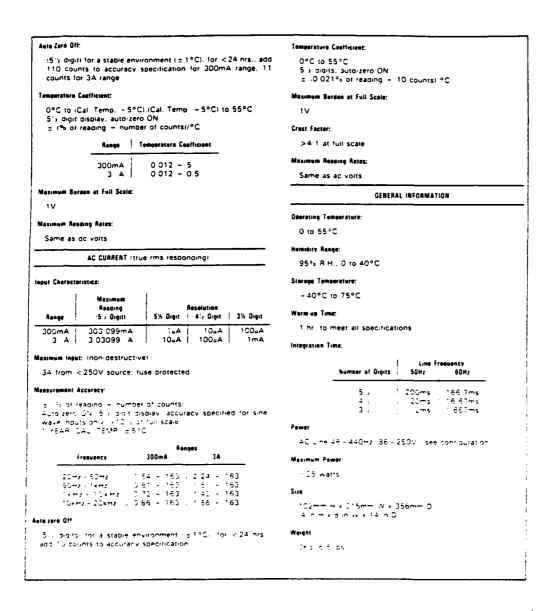


Figure 16. Technical Specifications for the Hewlett-Packard Model 3478A Multimeter. (continued)

TECHNICAL SPECIFICATIONS FOR THE HEWLETT-PACKARD MODEL 6825A BIPOLAR POWER SUPPLY/AMPLIFIER

GENERAL SPECIFICATIONS

Input Power:

104-127/208-254Vac (switchable), 48-63Hz, 1.0A, 150W

Meters:

Individual voltage and current meters. DC accuracy is 3% of full scale. AC accuracy is 5% of full scale with sinusoidal, 100Hz input.

Meter Ranges (DC):

±2.4V, ±24V/±0.24A, ±2.4A

Meter Ranges (AC):

1.6V (uncai), 16V rms/0 16A rms, 1.6A rms

Temperature Ratings:

Operating 0 to 55°C Storage -40 to +75°C.

Cooling:

Convection cooling is employed. The supplies have no moving parts.

Dimensions:

See outline diagram, Figure 2-1,

Weight:

18 lps. (8.2 kg.) net, 21 lbs. (9.5 kg.) shipping.

POWER SUPPLY SPECIFICATIONS

DC Output:

CONTROL CONTROLS DESIGNATE BUSINESS

Voltage and current spans indicate range over which output may be varied

X1 Range. -5V to +5V. 0 to 2.0A X4 Range -20V to +20V, 0 to 2.0A

Load Effect (Load Regulation):

Voltage load effect is given for a load current change equal to the current rating of the supply. Current load effect is given for a load voltage change equal to the voltage rating of the supply.

Voltage (X1 Range) | 0.01°. + 1mV Voltage (X4 Range) | 0.01°. + 5mV

Current 01% • 250µA

Source Effect (Line Regulation)

For a change in line voltage between 104 and 127Vac-208 and 254Vac at any output voltage and current within rating

Source Effect (Line Regulation) Continued:

Voltage (X1 Range): 01% + 2mV Voltage (X4 Range): 01% + 2mV Current: .01% + 250µA

PARD (Ripple and Noise):

Rms/p-p (20Hz to 20MHz) at any line voltage and under any load condition within rating.

Voltage (X1 Range): 1.5mV rms/4mV p-p Voltage (X4 Range): 5mV rms/15mV p-p Current: 3mA rms/10mA p-p

Temperature Coefficient:

Output change per degree Centigrade change in ambient following 30 minutes warm-up.

Voitage (X1 Ranger 01% + 35mV Voitage (X4 Ranger 01% + 1.5mV Current, .02% + 100µA

Drift (Stability):

Change in output (do to 20Hz) over 8 hour interval under constant line, load, and ampient following 30 minutes warm-up.

Voltage (X1 Range): .03% + 1mV (Pot wiper (umplettect may add 5mV)

Voltage (X4 Range): .03% + 5mV (Pot wider jump effect may add 50mV)

Current: 1% + 200µA (Pot wiper jump effect may

ad 1.5mA)

Load Effect Transient Recovery (Load Transient Recovery):

Time required for output voltage recovery to within the specified level of the nominal output voltage following a change in output current equal to the current rating of the supply.

100used is required for output voltage recovery within 20mV of nominal output voltage.

Resolution:

Typical output voltage or current change that can be obtained using front panel controls

Voltage (X1 Range) 10mV Voltage (X4 Range) 40mV

Output Impedance (Typical to 50kHz):

Approximated by a resistance in series with an inductance (constant voltage operation).

5m2 & 15uH

Figure 17. Technical Specifications for the Hewlett-Packard Model 6825A Bipolar Power Supply/Amplifier.

[Ref. 9: pp. 1.2-1.3]

POWER SUPPLY SPECIFICATIONS (Continued) Voltage Gain (High/Low Range): Fixed Amplifier (Inverting): 4X (high range)/ 1X How range DC Output Isolation: Supply may be floated at up to 300V above ground. Variable Gain (Non-Inverting): 0-8 (high range)/ 0-2 (low range) Remote Resistance Programming: Resistance Coefficient: Frequency Response (+1, -3dB at full output): Voltage (X1 Range): 2000Ω/V ± .1% Fixed Gain: dc - 40kHz Variable Gain: dc - 15kHz Voltage (X4 Range): 500Ω/V ± .1% Current: 5Ω/mA ± .1% Remote Programming Speed: Total harmonic distortion is .1% (maximum) at 100Hz 50µsec are required to change between 1% and 99% of and full output. the maximum + and - voltage limits. input impedance: Remote Programming Temperature Coefficient: 10KΩ (Typicai) Output change per degree Centigrade change in am-Fixed Gain Accuracy (at 100Hz): bient using an external control resistor (RF) at output voitage (VO) or current (IO). % T.C. Rp is the tem-Low Range (X1): 1% + 5mV High Range (X4) 1% + 2mV perature coefficient of the control resistance Rp. Voitage (X1 Range): .25mV + .007% (Vg) + Remote Resistance Programming Variable Gain (Ay): % T.C. RF (VO + 5) Voltage (X4 Range): 1mV + .007% (VO) + $\frac{10.24\times10^3\Omega}{10.24\times10^3\Omega}$, where K is the constant indi KRE % T.C. RF (VO + 20) Current: .016% (IO) + 33uA + % T.C. RE (IO) cated and Rip is the external control resistance. Sink Current Compliance: Ay at low range (X1): Maximum current that the supply can sink when connected to an active load. Ay at high range (X4) 10.24 x 103 +20v SOURCE Variable Gain Accuracy: Accuracy in high range at 100Hz using an external control resistance (RE) at output voltage (Vigit). Ris the accuracy of the control resistance Ris SOURCE 105 +182 Vg + 2.5mV Remote Voltage Control Coefficient: Fixed call amplitier mode, voltage coefficient Voltage X1 Range: Tivoit voit ± 15 Sink current is limited to a value ranging linear a from Voitage : X4 Hange: 4 voits voit ± 1% 2A 2 0V to 1A = 20V Externally applied voltages to output terminals in ex cess of 25V could damage the instrument. Variable gain amplifier mode (VOLTAGE control *UL v CIDCKWise - voltage opetficient

POWER AMPLIFIER SPECIFICATIONS

Voltage (X1 Range) 10V p-p

Voitage IX4 Range: 40V p.p.

Current: 2A peak

Output:

Figure 17. Technical Specifications for the Hewlett-Packard Model 6825A Bipolar Power Supply/Amplifier. (continued)

voltage Y1 Range: 2 voits voit ± 1%

Voltage: X4 Hanger: Bivoits;vr. 12, 1%.

Constant Current: collage coefficient (the following)

and power supply modes of operation

2 amperes voir ± 5 s

applies to variable gain amplifier, tixed gain amplifier

TECHNICAL SPECIFICATIONS FOR THE HEWLETT-PACKARD MODEL 59501B ISOLATED DAC/POWER SUPPLY PROGRAMMER

POWER SUPPLY PROGRAMMING D/A CONVERTER Programming Network Specifications: In the following spec-DC Output Voltage: Programmable in high or low ranges within the voltage limits shown below. Output mode is ifications. M represents the calibrated full scale value of the supply being programmed and P represents the actual prounipolar or bipolar and is selectable via rear panel switch. grammed output. Note that the full scale value (M) can be any value within the supply's output range and is calibrated High 0 to +.999 Volts with the 59501B programmed to its maximum high range Unipolar 0 to 9.99 Volts -10 to +9.98 Volts -1 to +.998 Volts output. Bipolar DC Output Current: 10mA Accuracy (Does not include power supply errors): Specified at 23°C ± 5°C. Resolution: High Low High Low Unipolar 10m V 1mV Unipolar .05%M + .25%P 01%M + .25%P 20mV 0.1%M + .25%P Bipoiar .02%M + .25%P Bipolar Accuracy: Specified at 23° C ± 5°C Isolation: 600Vdc between HP-IB data lines and output High Low terminals. .1% + 1mV 1% + 5mV Unipolar Temperature Coefficient: .1% + 10mV .1% + 2mV Bipolar .005%M/4C + .015%P/4C High .01%M/°C + .015%P.°C * Stability: Change in output over 8 hour interval under constant line, load, and ambient following a 30 minute Programming Resolution: High warm-up. Low Low High 0.1%M 0.01%M Unipolar 04% + .1mV 04% + 5mV Bipolar 0.2°6M 0.02° M .04% + 1mV .04% + .2mV Bipolar Programming Speed: D/A Conversion Time plus the Temperature Coefficient: programming speed of the power supply Low GENERAL High 019: "C + 1mV "C .01%, "C + 5mV/"C Input Power: Unit has ac power module which is settable .01%/°C + 1mV/°C 01%,~C + 5mV,~C to 100/120/220 240 vac +13% +5%, 46 53Hz 10VA Bipolar A 3-wire detachable line cord is supplied Zero Adjust: Plus or minus 250 millivolts. Temperature Range D/A Full Scale Adjust Plus or minus 5% Operating 0 to 55°C Storage -40 to 75°C Programming Speed. The time required for output to go from zero to 99% of programmed output change is $250 \mu sec$ Dimensions: (measured with resistive load connected to output terminals) (See Figure 2.1)

Figure 18. Technical Specifications for the Hewlett-Packard Model 59501B Isolated DAC/Power Supply Programmer. [Ref. 10: pp. 1-3]

* Stability is included in accuracy specification measure-

ments over the temperature range indicated.

Weight:

Net 1 82 kg -4 lb.1 Shipping 2.27 kg. 5 lb.1

TECHNICAL SPECIFICATIONS FOR THE IBM DATA ACQUISITION ADAPTER (ANALOG INPUT DEVICE)

Analog Input Device

The analog input device has the following characteristics:

Resolution

12 bits

Input Channels

4 differential

Input Ranges

Switch-selectable ranges: 0 to +10 volts (unipolar), -5 to +5 volts (bipolar), and -10 to +10 volts (bipolar).

Input Resistance

100 megohms minimum

Input Capacitance

200 picofarads maximum; measured at the distribution panel connector

Input Leakage Current

±300 nanoamperes maximum

Input Current

±4 milliamperes at maximum input

voltage

Digital Coding

Unipolar: binary. Bipolar: offset binary.

Safe Input Voltage

±30 volts maximum (power On or

Off)

Power Supply Rejection

±1/2 LSB maximum change full

scale calibration

Integral Linearity Error

±1 LSB maximum

Figure 19. Technical Specifications for the IBM Data Acquisition Adapter (Analog Input Device).

[Ref. 11: pp. 116-118]

Differential Linearity Error

±1/2 LSB maximum

Differential Linearity Stability

±5 ppm/°C maximum; guaranteed monotonic

Gain Error

±0.1% maximum between ranges. Any range adjustable

to zero.

Gain Stability

±32 ppm/°C of FSR

maximum

Common-Mode Input Range

±11 volts maximum

Common-Mode Rejection

72 dB minimum ratio (signal within common-mode range)

Unipolar Offset Error

Adjustable to zero

Unipolar Offset Stability

±24 ppm '°C of FSR

maximum

Bipolar Offset Error

Adjustable to zero

Bipolar Offset Stability

±24 ppm '°C of FSR

maximum

Figure 19. Technical Specifications for the IBM Data Acquisition Adapter (Analog Input Device). (continued)

Settling Time

For channel acquisition: 20

microseconds maximum to $\pm 0.1\%$

of the input value.

Conversion Time

35 microseconds maximum

Throughput to Memory

15,000 conversions per second,

minimum

'A/D convert enable'

Input Impedance

One LS TTL load plus 10-kilohm

pull-up resistor

'A/D convert out'

Fanout

10 LS TTL loads or 2 standard

TTL loads

Figure 19. Technical Specifications for the IBM Data Acquisition Adapter (Analog Input Device). (continued)

APPENDIX B

OPERATIONAL AND CALIBRATION PROCEDURES

This appendix contains specific detailed instructions for operation and calibration of the solar simulator laboratory and the automated solar cell test program. The operational procedures are written in a step-by-step format. Since the use of the Kratos light source requires caution, it is recommended that they be followed in order.

A. OPERATIONAL PROCEDURES

- 1. One hour prior to test-
- (a) Turn on temperature control circulator and select desired test temperature.
 - (b) Turn on circulator pump housing vent fan.
 - (c) Turn on Omega digital temperature display.
 - 2. One half hour prior to test-
- (a) Turn on water circulating pump for the IR filter.
 - (b) Turn on upper and lower lamp housing vent fans.
 - (c) Start Kratos light source power supply.
- (d) Disconnect DAC adapter. This step is recommended due to the possibility of damage to the DAC adapter card by the 40 KV R.F. pulse needed to start the Kratos lamp.
 - (e) Plug in Kratos lamp starter unit.

- (f) Put on safety goggles (filters ultraviolet radiation to protect the eyes).
- (g) Activate the toggle switch on the starter unit to the `start' position. Hold for no longer than 10 sec. The switch is a momentary contact type and will spring release. The lamp should start immediately, however if it is hot, a cool down period is required prior to restart. If difficulties are encountered consult the Kratos literature.
 - (h) Disconnect the Kratos lamp starter unit.
 - (i) Turn on HP 59501B power supply programmer.
 - (j) Turn on HP 6825A bipolar power supply.
 - (k) Turn on HP 3478A digital multimeter.
 - 3. Fifteen minutes prior to test-
- (a) Turn "cool-on" switch on temperature control circulator to on position.
 - (b) Turn on IBM PC/XT and HP 7475 plotter.
- (c) Plug in DAC adapter connection cable. Arrow points upward.
 - 4. Test time-
 - (a) Start vacuum pump.
- (b) Place test/setup switch on test module to
 "setup" position.
- (c) Using a secondary standard, verify Air Mass Zero solar intensity, by comparing the secondary standard output voltage to that voltage supplied with the standard

cell documentation. Adjust lamp current or cell position as necessary to achieve proper voltage.

- (d) Place solar cell on test block using plastic tweezers.
- (e) Start test program by typing "BASICA CORPROG" to the right of the cursor C>, and then follow the user friendly directions.
- (f) When running the calibration routine the Eppley standard cell voltage should be 1.0188 +/- .0002 volts.

5. Post test-

- (a) Secure the Kratos lamp power supply.
- (b) Secure the vacuum pump.
- (c) Secure the temperature control circulator pump.
- (d) Turn off the HP 6825A power supply, HP 59501B power supply programmer, computer, and plotter.
- (e) Leave all vent fans and the IR filter water circulator running for at least one (1) hour following Kratos shutdown.

B. Calibration Procedures

1. Discussion

Two areas of the test system require periodic calibration. These are: the A/D channels used for voltage measurements and the programmable power supply output.

2. Analog to Digital Converter Calibration

As discussed in Chapter III, selecting menu routine number l in the test program calculates the voltage

offsets required for calibration of the two analog to digital converter channels. Since the two offsets calculated are used throughout the cell test routine, this routine must be run prior to commencement of a test session. The calibration of menu routine number 1 needs to be run only once upon entering the program, however, since the values of the offsets are stored in the variables Cl and C2. Therefore, the calibration of the A/D card is assured each time the routine is run.

3. Power Supply Programming

Programming of the power supply to a given output is accomplished by calculating a specific digital code and transmitting this code over the IEEE 488 GPIB bus to the HP 59501B/6825A power supply combination which outputs the appropriate analog voltage. For this application the code is a number between 1000 and 1999. The total number of programming steps available for use is 999. A code of 1999 corresponds to a maximum output and a code of 1000 corresponds to a minimum output. The resolution of the power supply may then be determined by dividing the total voltage swing by the total number of programming steps. Algebraically,

$$[V_{\text{max}} + V_{\text{min}}] / 999 = \text{resolution (volts/step)}$$
 (9)

For this application the maximum open circuit voltage for any one gallium arsenide cell used was 1.001 volts. Thus V_{max} was initially chosen to be 1.010 volts. The power supply literature states that the minimum resolution possible is 1.5 millivolts/step which yields a V_{min} of -0.4885 volts (1.01 - 999 x 0.0015). Adjustments were made to the power supply to yield a V_{max} of 1.0098 volts and a V_{min} of -0.4629 volts and a resolution of

$$[1.0098 + 0.4629] / 999 = 1.47 \text{ mV/step}$$
 (10)

which is greater than the specifications would suggest. For these values of $V_{\mbox{max}}$ and $V_{\mbox{min}}$, a zero voltage output was achieved with a code of 1314.

Any desired voltage may then be coded by the relation:

code = [Voltage desired] / $[V_{max} + V_{min}] \times 999 + zero$ offset (11)

$$code = V_{des} / 1.4727 \times 999 + 1314$$
 (12)

or

$$code = V_{des} \times 678.346 + 1314$$
 (13)

where 678.346 is the constant multiplier (c.m.). (See line 3280 of the program in Appendix B).

a. Power Supply Calibration Procedures

Power supply calibration is required when the voltage programmed is not equal to the voltage output. This condition is evidenced by the output voltage failing to be within 1.5 mV of the cell $V_{\rm OC}$. A statement to this effect is displayed to the user by line number 3500 in the cell test subroutine. The following steps detail the calibration procedure required:

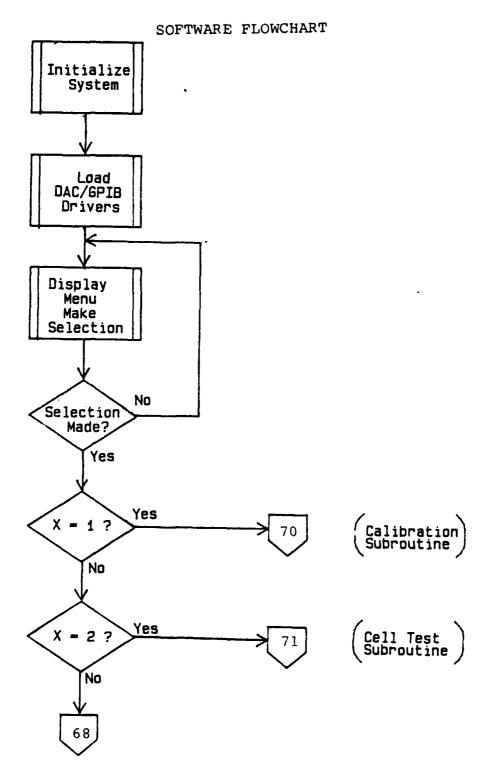
- (1) Connect the power supply outputs to a digital voltmeter with a minimum resolution of 10 microvolts
- (2) Enter the GPIB sub-directory on the IBM PC/XT.
- (3) Type IBIC [return] at the C> prompt (this initiates a GPIB program that allows keyboard input to a GPIB programmable device).
- (4) Type IBFIND psupp [return] at the IBIC> prompt.
- (5) Type IBWRT "1999" at the psupp: prompt. This drives the power supply to maximum output. The voltage should be 1.0098 volts (+/- 100 microvolts). Record the voltage.
- (6) Type IBWRT "1000". This drives the power supply to -0.4629 volts (+/- 100 microvolts). Record the voltage.

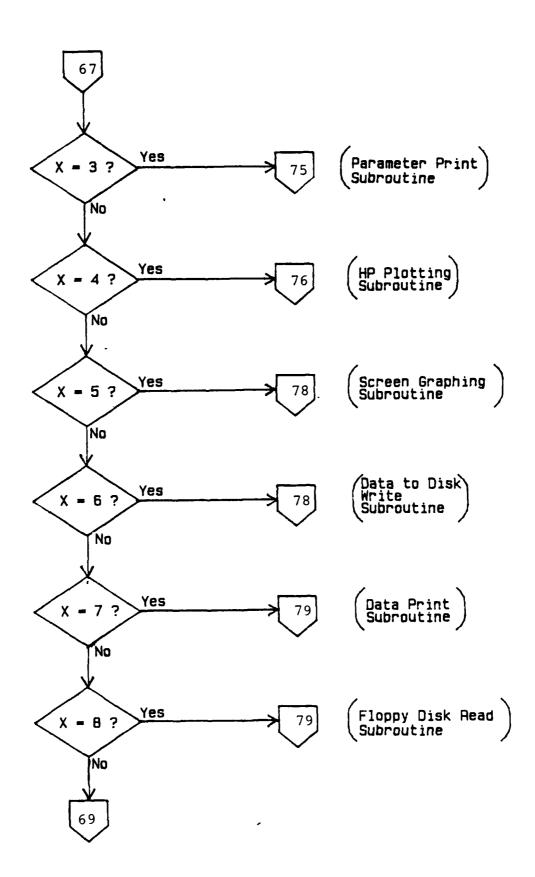
- (7) Type IBWRT "1314". The output should now be -0.03 millivolts (+/- 10 microvolts). Record the voltage.
- (8) If the above specifications are not met, then the HP 59501B power supply full scale adjust and zero adjust potentiometers must be adjusted to regain the required precision. This procedure entails frequent use of the IBWRT command and may require a significant amount of time.
- (9) Upon completion of steps (5) (7), recalculate the constant multiplier (c.m.) in equation (13), or

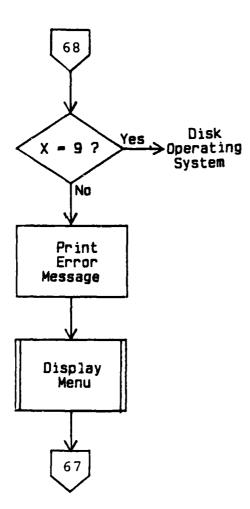
c.m. = 999 /
$$[v_{max} + v_{min}]$$

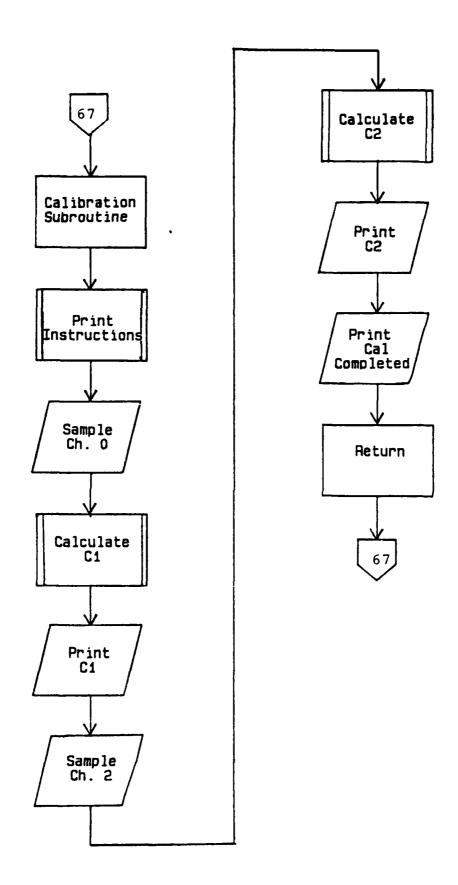
- (10) Insert this value in line 3280 of the test program.
- (11) Calculate 70 % of the c.m. and insert this value in line 3290.
- (12) Calculate 3 % of the c.m. and insert this value in line 3300.
- b. It should be noted that the HP 59501B/6825A combination is exceptionally stable after a 30 minute warm-up period, and this calibration procedure is not often required.

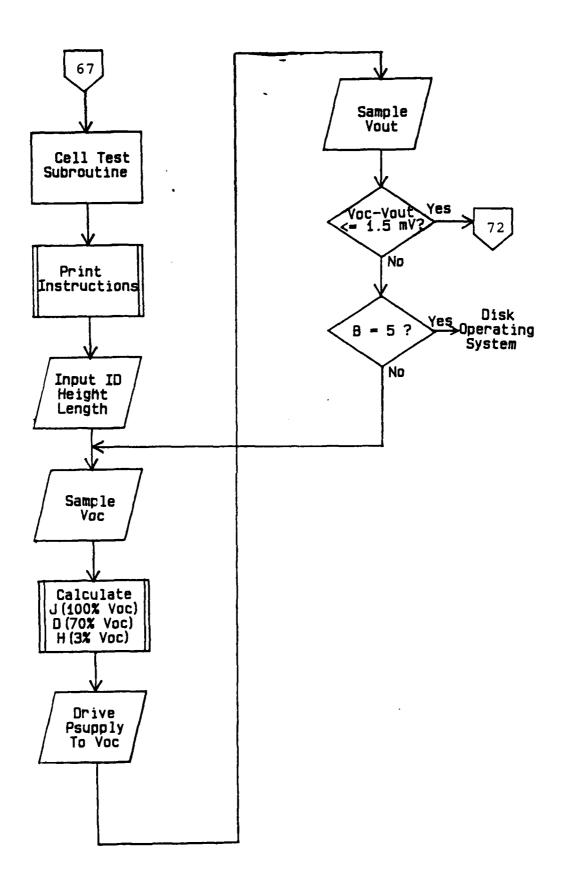
APPENDIX C

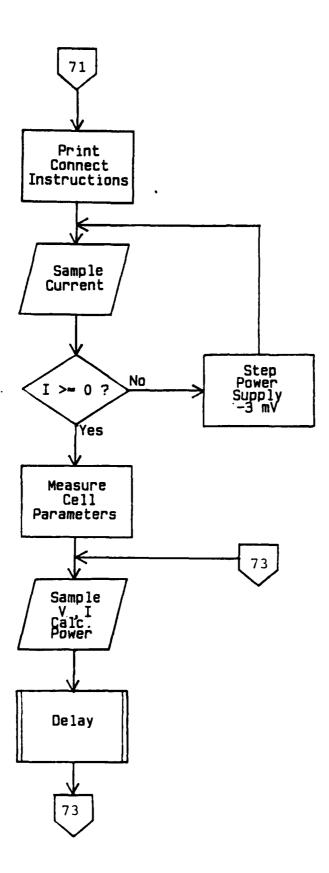








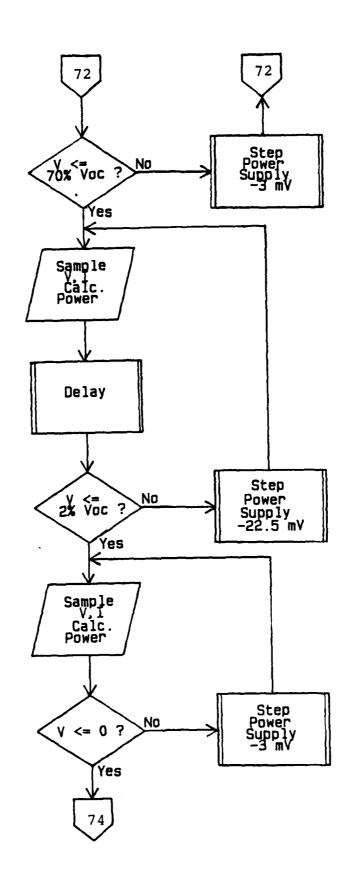


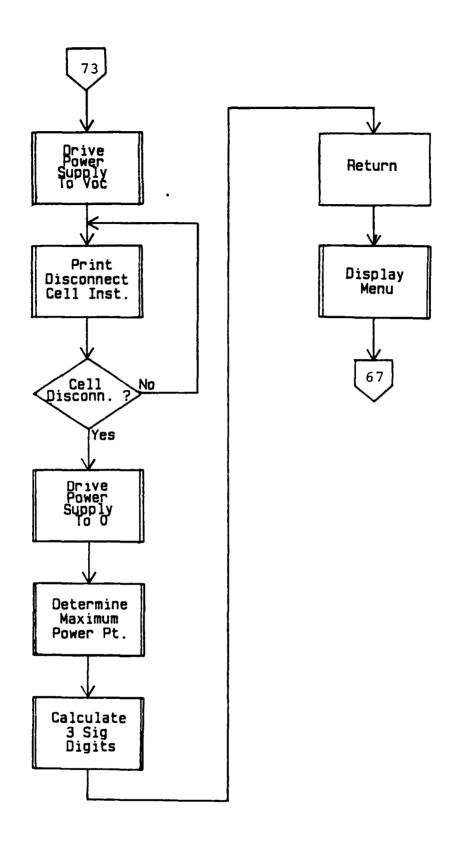


CHANGE PARTY OF

CONTRACTOR OF STREET

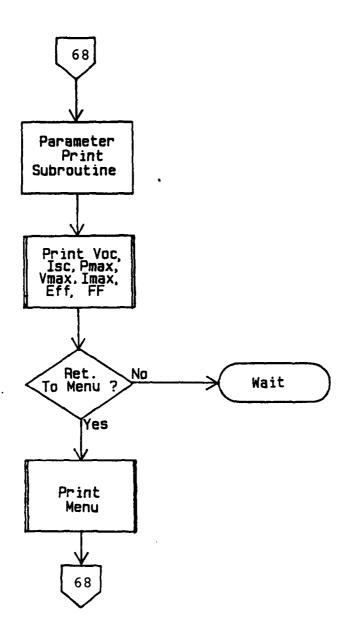
coloridat essection this cooks consist the



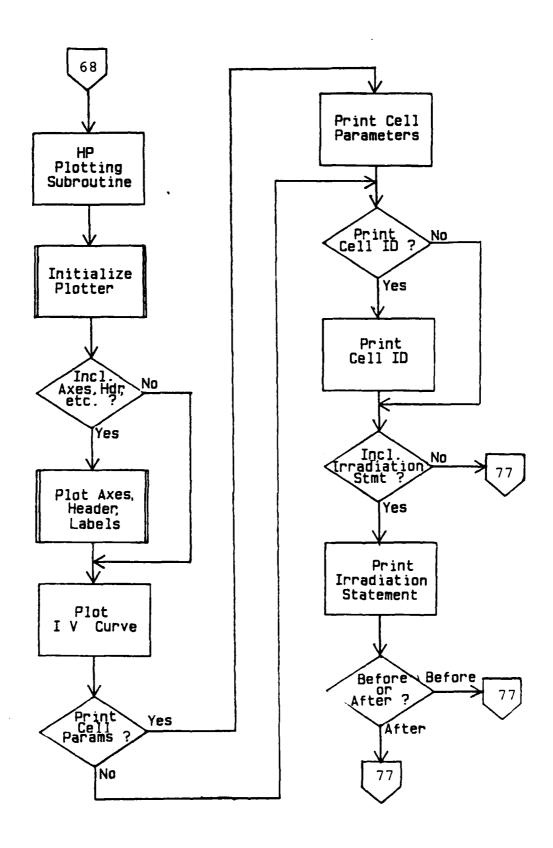


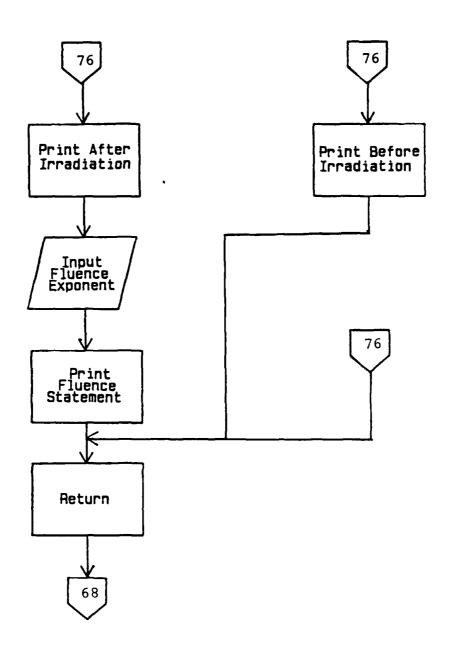
DEZZE MAGESTOS PAGESTAS DZIOLAJA DZIOLAJA SUSSIANI

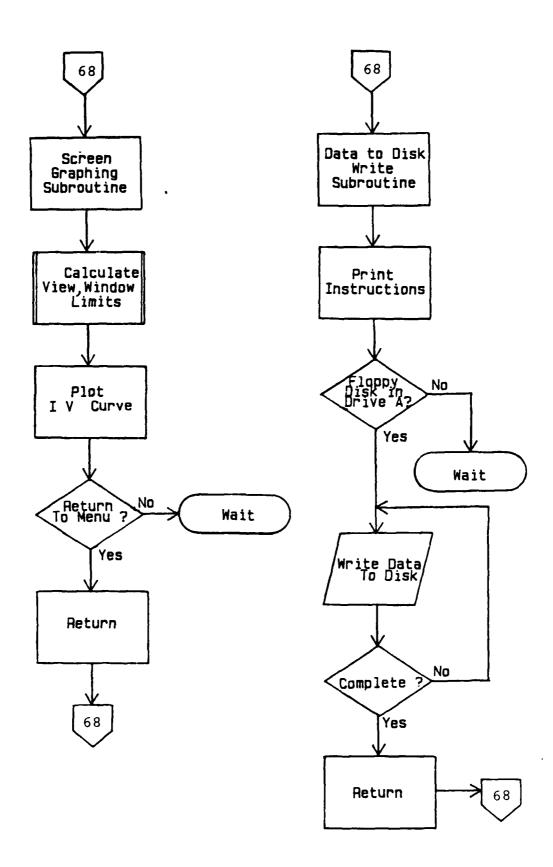
CACCACACA COCCACACA MEDICINAL BESTERSON PARAMETERS IN



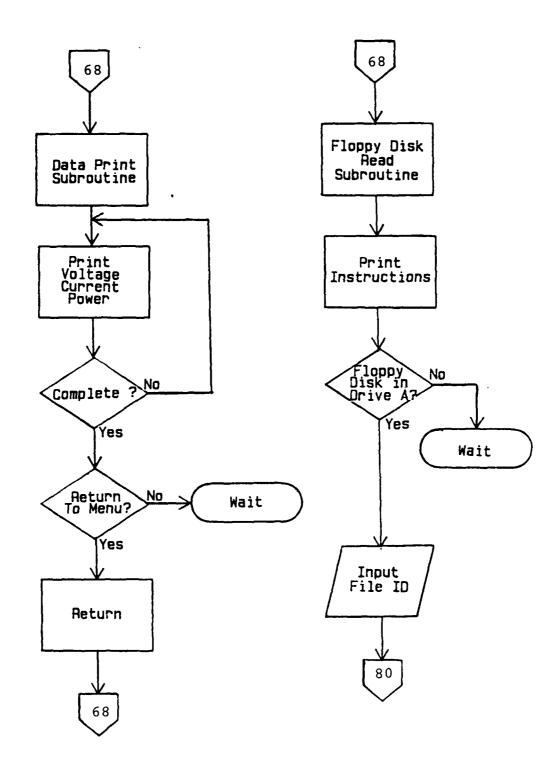
Contract Contract to Contract Distriction

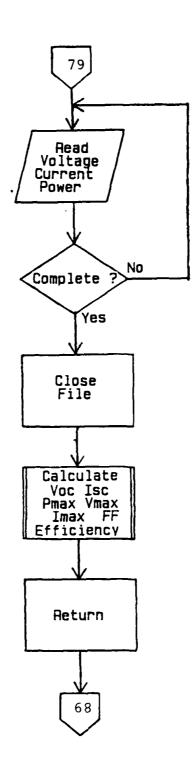






and apprinted between the second





SECTIONS INVOICED DEPONDE EXCENSE BESSE

APPENDIX D

SOLAR CELL TEST PROGRAM

100	•	*************	*
110	•	* CORPROG.BAS	*
120	•	*	*
130	^	* BY	*
140	•	*	*
160	•	* DON W. GOLD	*
180	•	*	*
190	•	* THIS PROGRAM WAS WRITTEN TO PROVIDE AN AUTO-	*
200	•	* MATIC METHOD FOR DETERMINING SOLAR CELL PARA-	*
210	•	* METERS. IT IS MENU DRIVEN AND REQUIRES THE	*
220	•	* USER TO PROVIDE AN IBM DATA ACQUISITION CARD,	*
230		* A GPIB INTERFACE CARD, AN IEEE-488 CAPABLE	*
240		* MULTIMETER, AND AN IEEE-488 CAPABLE BIPOLAR	*
250	,	* POWER SUPPLY. THE GRAPHICS PORTION OF THE	*
260	,	* PROGRAM IS WRITTEN FOR AN HP 7475A PLOTTER	*
270	,	* AND THE CONNECTION DIAGRAM FOR THE SYSTEM MAY	*
280 290	,	* MAY BE FOUND IN CH 2, FIGURE (3). THE GPIB * DRIVER IS REPRODUCED FROM REF 12 AND THE DAC	*
300	,	* DRIVER IS REPRODUCED FROM REF 12 AND THE DAC * DRIVER IS REPRODUCED FROM REF 13. THEIR	*
310	,	* INCLUSION IN THE PROGRAM IS REQUIRED TO	*
320	•	* OPERATE THE RESPECTIVE DEVICE.	*
330	•	***************	*
340	•		
350	•		
360	•	**************************************	*
370	•		
380		CLEAR ,59309! IBM BASICA Declaration	ıs
390		IBINIT1 = 59309!	
400		IBINIT2 = IBINIT1 + 3	
410		BLOAD "bib.m", IBINIT1	
420		CALL IBINIT1(IBFIND, IBTRG, IBCLR, IBPCT, IBSIC,	
		IBLOC, IBPPC, IBBNA, IBONL, IBRSC, IBSRE, IBRSV, IBPAD,	
430		<pre>IBSAD, IBIST, IBDMA, IBEOS, IBTMO, IBEOT, 1BRDF, 1BWRTF CALL IBINIT2(IBGTS, IBCAC, IBWAIT, IBPOKE, IBWRT,</pre>	`)
430		IBWRTA, IBCMD, IBCMDA, IBRD, IBRDA, IBSTOP, IBRPP, IBRS	· D
		IBDIAG, IBXTRC, IBRDI, IBWRTI, IBRDIA, IBWRTIA, IBSTA%	
		IBERR*, IBCNT*)	,
440		REM Optionally include the following declaration	. 5
		in your program.	
450		REM They provide appropriate mnemonics by which	
460		REM to reference commonly used values. Some	
		mnemonics (GET%, ERR%, END%, ATN%) are preceded	
470		by "B" in order to distinguish them from BASICA	
480		REM keywords.	
490		REM	

```
500
         REM GPIB Commands
510
         UNL% = &H3F
                             GPIB unlisten command
520
         UNT% = &H5F
                             GPIB untalk command
530
         GTL% = &Hl
                             GPIB go to local
540
         SDC% = &H4
                             GPIB selected device clear
550
         PPC% = &H5
                             GPIB parallel poll configure
560
         BGET% = &H8
                             GPIB group execute trigger
570
                            GPIB take control GPIB local lock out
         TCT% = &H9
580
         LLO% = &Hll
590
         DCL% = &H14
                             GPIB device clear
                             GPIB ppoll unconfigure
600
         PPU% = &H15
610
         SPE% = &H18
                            GPIB serial poll enable
                             GPIB serial poll disable
620
         SPD% = \&H19
                             GPIB parallel poll enable
630
         PPE% = &H60
640
         PPD% = &H70
                             GPIB parallel poll disable
650
         REM
660
         REM GPIB status bit vector
         REM global variable IBSTA% and wait mask
670
680
         BERR% = &H8000
                            Error detected
690
         TIMO% = &H4000
                             Timeout
700
         BEND% = &H2000
                             EOI or EOS detected
710
         SROI% = \&H1000
                             SRQ detected by CIC
720
         RQS% = &H800
                             Device needs service
730
         CMPL% = &H100
                             I/O completed
740
         LOK% = &H80
                             Local lockout state
750
         REM% = &H40
                            Remote state
760
         CIC% = \&H20
                             Controller-In-Charge
770
                            Attention asserted
         BATN% = &HIO
780
         TACS% = &H8
                            Talker active
790
         LACS% = &H4
                            Listener active
800
         DTAS% = &H2
                            Device trigger state
                            Device clear state
810
         DCAS% = &H1
820
         REM
830
         REM Error message; returned in variable IBERR%
                            DOS error
840
         EDVR% =
                   0
850
         ECIC% =
                   1
                             Function req GPIB-PC to be CIC
860
         ENOL%
                            Write function
                                              detected
         Listeners
870
         EADR% =
                   3
                             Interface board
                                               not
         addressed correctly
880
         EARG% =
                   4
                             Invalid argument to function call
890
         ESAC% =
                   5
                             Function requires GPIB-PC to be
         SAC
         EABO% =
900
                   6
                             I/O operation aborted
                   7
910
         ENEB% =
                             Non-existent interface board
920
         EOIP% =
                   10
                             1/0
                                 operation started before
         previous op completed
930
         ECAP% = 11
                            No capability for operation
940
         EFSO% = 12
                             File system operation error
950
         EBUS% = 14
                            Command error during device call
960
         ESTB% = 15
                             Serial poll status byte lost
```

THE RESIDENCE OF THE PROPERTY OF THE PERSON OF THE PERSON

```
ESRO% = 16 SRQ remains asserted
970
980
          REM
990
          REM EOS mode bits
                            Eight bit compare
1000
          BIN% = &H1000
1010
          XEOS% = &H800
                            Send EOI with EOS byte
                           Terminate read on EOS
1020
          REOS% = \&H400
1030
          REM
1040
          REM Timeout values and meanings
                         Timeout of 10 s (ideal)
1050
          T10S% = 13
1060
          REM
1070
          REM Miscellaneous
                            Parallel Poll sense bit
1080
          S% = \&H8
                          Line feed character
          LF% = &HA
1090
1100
          REM
1110
          REM Application program variables passed to
          REM GPIB functions
1120
1130
          REM
                                  command buffer
read data buffer
1140
          CMD$ = SPACE$(10)
1150
1160
          RDS = SPACES(255)
                                 write data buffer
          WRT$ = SPACE$(255)
                                  board name buffer
1170
          BNAME$ = SPACE$(7)
                                  'board or device name
1180
          BDNAME$ = SPACE$(7)
          buffer
1190
          FLNAME$ = SPACE$(50)
                                  file name buffer
         ***********END OF GPIB DRIVER***********
1200
1210
1220
          DEF SEG = 0
1230
          DATA.SEGMENT = PEEK(&+511) * 256 + PEEK(&+510)
1240 BDNAME$ = "PSUPP"
1250 DEF SEG = DATA.SEGMENT 'define memory location
1260 CALL IBFIND (BDNAME$, PSUPP%)
1270 \text{ BDNAME} = "DMM"
1280 CALL IBFIND (BDNAME$, DMM%)
1290 B$ = "F5N5Z0T1" 'set dmm to 5 1/2 digits,DC current
1300 CALL IBWRT(DMM%,B$) 'remote operation
1310
1320
1330 1
         ******DAAC BASICA HEADER*****
1340
1350
     NAME: Data Acquisition And Control (DAAC)
1360
             HEADER for BASICA
1370
1380 'FILE NAME: DACHDR.BAS
1390
1400 'DOS DEVICE NAME: DAAC
1410
1420 'RESERVED FUNCTION NAMES:
                AINM, AINS, AINSC, AOUM, AOUS,
1430
1440
                BINM, BINS, BITINS, BITOUS, BOUM, BOUS,
1450
                CINM, CINS, CSET, DELAY
1460 'RESERVED DEF SEG VALUE NAME:
```

```
1470
1480
     NAMES DEFINED AND USED BY HEADER:
1490
                 ADAPT%, AI, COUNT, FOUND%,
1500
                .HNAME$, SG%, STAT%
1510
1520 \text{ FOUND} = 0
1530 \text{ SG}\% = \&\text{H2E}
     Start searching the interrupt vectors until you find
1540
     one that points to the DAAC device driver.
     Do a DEF SEG to that segment.
1570 WHILE ((SG% \leftarrow &H3E) AND (FOUND% = 0))
1580
            DEF SEG = 0
1590
            DSEG = PEEK(SG%) + PEEK(SG% + 1) * 256
1600
            DEF SEG = DSEG
            HNAME $=""
1610
1620
            FOR AI=10 TO 17
1630
                    HNAME$ = HNAME$ + CHR$(PEEK(AI))
1640
            NEXT AI
1650
            IF HNAME$ = "DAAC" AND PEEK(18)+ PEEK(19) <> 0
            THEN FOUND  = 1 
1660
            SG% = SG% + 4
1670
            WEND
                FOUND% = 0 THEN PRINT "ERROR: DEVICE DRIVER
1680
            IF
                        NOT FOUND" : END
            DAC.COM
1690
             Initialize all function name variables for calls
1700
            'to access the device driver.
                  = PEEK(&H13) * 256 + PEEK(&H12)
1710
            AINM
                  = PEEK(&H15) * 256 + PEEK(&H14)
1720
            AINS
            AINSC = PEEK(&H17) * 256 + PEEK(&H16)
1730
                  = PEEK(\&H19) * 256 + PEEK(\&H18)
1740
            AOUM
                  = PEEK(\&HlB) * 256 + PEEK(\&HlA)
1750
            AOUS
                  = PEEK(\&HlD) * 256 + PEEK(\&HlC)
1760
            BINM
1770
            BINS
                  = PEEK(\&H1F) * 256 + PEEK(\&H1E)
            BITINS = PEEK(\&H21) * 256 + PEEK(\&H20)
1780
1790
            BITOUS = PEEK(&+23) * 256 + PEEK(&+22)
                   = PEEK(&H25) * 256 + PEEK(&H24)
1800
            BOUM
                   = PEEK(&H27) * 256 + PEEK(&H26)
1810
            BOUS
                   = PEEK(\&H29) * 256 + PEEK(\&H28)
1820
            CINM
1830
            CINS
                   = PEEK(\&H2B) * 256 + PEEK(\&H2A)
                   = PEEK(\&H2D) * 256 + PEEK(\&H2C)
1840
            CSET
1850
            DELAY
                   = PEEK(\&H2F) * 256 + PEEK(\&H2E)
1860 DIM V%(200)
1870 DIM N$(2000)
1880 DIM VOLT(1000)
1890 DIM CURR(1000)
1900 DIM PMAX(1000)
1910 DIM VIN%(20)
1920 \ ZS = SPACES(14)
1925 \text{ Cl} = -3
1930 'initialize global variables
1940 \text{ DEV}\% = 9
```

SSSSI SSSSSS KOKKOS JOSEDJE VEDDEDI ZVIDODE

```
1950 \text{ ADAPT} = 0
1960 \text{ COUNT} = 1
1970 \text{ STAT} = 0
1980 DEF SEG = DSEG 'define memory location
1990 CALL DELAY (ADAPT%, COUNT, STAT%)
2000
         *******END OF DAAC BASICA HEADER***********
2010
2020
2030 CLS
2040 SCREEN 0
2050 CLS
2060 BEEP
2070 PRINT "THE FOLLOWING OPTIONS ARE AVAILABLE FOR USE WITH
2080 PRINT "PROGRAM. SELECT A NUMBER AND THEN ENTER RETURN.
    NUMBER"
2090 PRINT "1 MUST BE SELECTED BEFORE NUMBER 2."
2100 PRINT
2110 PRINT "1. RUN CALIBRATION ROUTINE."
2120 PRINT
2130 PRINT "2. RUN SOLAR CELL PARAMETER TEST."
2140 PRINT
2150 PRINT "3. PRINT SOLAR CELL PARAMETERS ON SCREEN."
2160 PRINT
2170 PRINT "4. PLOT I-V CURVE ON HP 7845 PLOTTER."
2180 PRINT
2190 PRINT "5. PLOT I-V CURVE ON RGB MONITOR."
2200 PRINT
2210 PRINT "6. WRITE I-V DATA TO FLOPPY DISK."
2220 PRINT
2230 PRINT "7. PRINT I-V DATA ON SCREEN."
2240 PRINT
2250 PRINT "8. READ I-V DATA FROM FLOPPY DISK."
2260 PRINT
2270 PRINT "9. EXIT TO SYSTEM."
2280 PRINT
2290 INPUT ">",X
2300 PRINT
2310 IF X <> 1 THEN 2340
2320 GOSUB 2620 'go to calibration subroutine
2330 GOTO 2040
                  return to menu
2340 IF X <> 2 THEN 2370
2350 GOSUB 3000 'go to cell test subroutine
2360 GOTO 2040
2370 IF X <> 3 THEN 2400
2380 GOSUB 6140
                  'go to parameter print subroutine
2390 GOTO 2040
2400 IF X <> 4 THEN 2430
2410 GOSUB 5170 'go to HP plotting subroutine
2420 GOTO 2040
2430 IF X <> 5 THEN 2460
```

```
'go to screen graphing subroutine
2440 GOSUB 5030
2450 GOTO 2040.
2460 IF X <> 6 THEN 2490
2470 GOSUB 5990 'go to disk write subroutine
2480 GOTO 2040
2490 IF X <> 7 THEN 2520
2500 GOSUB 4910 'go to data print subroutine
2510 GOTO 2040
2520 IF X <> 8 THEN 2550
2530 GOSUB 6320 'go to floppy disk read subroutine
2540 GOTO 2040
2550 IF X <> 9 THEN 2570
2560 SYSTEM
2570 PRINT "ANY INPUT MUST BE FROM 1 TO 9"
2590 INPUT "PRESS ENTER TO RETURN TO MENU >",C$
2600 GOTO 2040
2610
         ***** CALIBRATION SUBROUTINE *********
2620
2630
2640 CLS
2650 PRINT "COMMENCING CALIBRATION ROUTINE"
2660 BEEP
2670 PRINT
2680 PRINT "CONNECT STANDARD CELL TO TEST MODULE AND PLACE"
2690 PRINT "CAL/RUN SWITCH TO CAL POSITION. ENTER THE VALUE"
2700 INPUT "OF THE STANDARD CELL VOLTAGE WHEN PROMPTED >", A
2710 PRINT
2720 \text{ CHAN}% = 0: CTRL% = 0: MODE% = 0
2730 STOR% = 0: COUNT = 20: RATE = 500 'sample @ 500 HZ
2740 \text{ STAT} = 0
2750 DEF SEG = DSEG
2760 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT',
     RATE, VIN% (0), STAT%)
2770 \text{ VIN} = 0
2780 \text{ FOR G} = 0 \text{ TO } 19
         VIN = VIN%(G) + VIN 'average 20 samples ch0
2790
2800 NEXT G
2810 Cl = VIN/8192 - A 'calculate constant Cl
2820 PRINT "C1=":C1
2830 CHAN% = 2: CTRL% = 0: MODE% = 0
2840 STOR% = 0: COUNT = 20: RATE = 500
                                          'sample @ 500 HZ
2850 \text{ STAT}\% = 0
2860 DEF SEG = DSEG
2870 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, VIN% (0), STAT%)
2880 \text{ VIN} = 0
2890 \text{ FOR G} = 0 \text{ TO } 19
         VIN = VIN%(G) + VIN 'average 20 samples ch2
2900
2910 NEXT G
2920 C2 = VIN/8192 - A 'calculate constant C2
```

THE PROPERTY OF THE PROPERTY OF THE PARTY OF

```
THESE VALUES SHOULD BE NEAR 5.0"
2930 PRINT "C2=";C2;"
2940 BEEP
2950 PRINT "CALIBRATION COMPLETED. PLACE THE CAL/RUN"
2960 PRINT "SWITCH IN THE RUN POSITION. PRESS ENTER TO"
2970 INPUT "RETURN TO MENU >",C$
2980 RETURN
2990
         ***** CELL TEST SUBROUTINE *************
3000
3002 CLS
3004 \text{ IF Cl} = -3 \text{ THEN } 3008
3006 GOTO 3020
3008 INPUT "RUN CALIBRATION ROUTINE FIRST.PRESS RETURN >",C$
3010
3020 CLS
3030 BEEP
3040 INPUT "INPUT THE CELL IDENTIFICATION NUMBER >", CELL$
3050 PRINT
3060 INPUT "INPUT THE CELL WIDTH IN CENTIMETERS >", W
3070 PRINT
3080 INPUT "INPUT THE CELL LENGTH IN CENTIMETERS >",L
3090 BEEP
3100 PRINT
3110 PRINT "MOVE SWITCH TO `SETUP' POSITION AND INSERT CELL"
3120 PRINT "IN FIXTURE. SELECT N/P OR P/N TYPE AS
     APPROPRIATE."
3130 INPUT "PRESS ENTER KEY WHEN DONE >",C$
3140 CLS
3150 B = 0 'count variable
3160 'measure cell open circuit voltage
3170 \text{ CHAN}% = 0: CTRL% = 0: MODE% = 0
3180 \text{ STOR}% = 0: COUNT = 20 : RATE = 500
                                            'sample @ 500 hz
3190 \text{ STAT}\% = 0
3200 DEF SEG =DSEG
3210 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, VIN% (0), STAT%)
3220 \text{ VIN} = 0
3230 \text{ FOR G} = 0 \text{ TO } 19
         VIN = VIN%(G) + VIN 'average 20 samples ch0
3250 NEXT G
3260 \text{ VOC} = \text{VIN}/8192 - \text{C1}
3270 PRINT "OPEN CIRCUIT VOLTAGE = "; VOC
3280 J = INT(VOC * 678.3459 + 1314)
3290 D = INT(VOC * 474.842 + 1314)
3300 H = INT(VOC * 20.35 + 1314)
3310 N$(J) = STR$(J): N$(J) = RIGHT$(N$(J),4)
3320 DEF SEG = DATA.SEGMENT
3330 CALL IBWRT (PSUPP%, N$(J))
     'determine value of power supply output
3350 CHAN% = 2: CTRL% = 0: MODE\% = 0
3360 STOR% = 0: COUNT = 10: RATE = 500 'sample @ 500 HZ
3370 \text{ STAT} = 0
```

```
3380 DEF SEG = DSEG
3390 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, VIN% (0), STAT%)
3400 \text{ VIN} = 0
3410 \text{ FOR G} = 0 \text{ TO } 9
3420
         VIN = VIN%(G) + VIN 'average 10 samples ch2
3430 NEXT G
3440 VOUT = VIN/4096 - C2 'calculate output voltage
3450 PRINT "POWER SUPPLY OUTPUT = "; VOUT
3460 IF ABS(VOC - VOUT) <= .0015 THEN 3580
3470 B = B + 1
3480 \text{ IF B} >= 5 \text{ THEN } 3500
3490 GOTO 3170 'reinitialize setup
            "VOUT NOT WITHIN 1.5 MILLIVOLTS OF VOC.
3500 PRINT
     PERFORM POWER SUPPLY CALIBRATION AS PER APPENDIX C".
3510 BEEP
3520 SYSTEM
3530
3540 1
         *******END INITIALIZATION ROUTINE**********
3550
3560 1
         *******CONNECT CELL TO TEST CIRCUIT*********
3570
3580 BEEP
3590 BEEP
3600 PRINT "MOVE SWITCH TO `TEST' POSITION, AND PRESS"
3610 INPUT "ENTER TO VERIFY ZERO CURRENT >",C$
3620 DEF SEG = DATA.SEGMENT
3630 CALL IBRD(DMM%, Z$) 'get current from dmm
3640 \text{ ITEST} = VAL(Z\$)
3650 IF ITEST >= 0 THEN 3700 'verify zero or positive
current
3660 J = J-1
3670 \text{ N}(J) = \text{STR}(J) : \text{N}(J) = \text{RIGHT}(\text{N}(J), 4)
3680 CALL IBWRT(PSUPP%,N$(J))
3690 GOTO 3620
3700 BEEP
3710 INPUT "PRESS ENTER TO BEGIN TEST >",C$
3720
3730
         ******PARAMETER MEASURING LOOP************
3740
3750 CLS
3760 PRINT "NOW MEASURING CELL PARAMETERS. WAIT 1 MINUTE
     FOR PROMPT."
3770 PRINT
3780 I = 0
3790 FOR M = J TO D STEP -2 'measure from Voc to 70% Voc
3800 I = I + 1
3810 \text{ CHAN}% = 0: CTRL% = 0: MODE% = 0
3820 STOR% = 0: COUNT = 5: RATE = 500 'sample @ 500 hz
3830 \text{ STAT}\% = 0
3840 DEF SEG = DSEG
```

(cess) excesses excesses entropied ())(in)) definition

```
3850 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, V% (0), STAT%)
        VTOT = 0
3860
        FOR X = 0 TO 4
3870
                VTOT = V%(X) + VTOT average 5 samples ch0
3880
3890
        NEXT X
3900 DEF SEG = DATA.SEGMENT
3910 CALL IBRD(DMM%,Z$)
                          'calculate current
3920 \text{ CURR}(I) = VAL(Z\$)
3930 VOLT(I) = VTOT/2048 - Cl 'calculate voltage
3940 PMAX(I) = VOLT(I) * CURR(I)
                                   'calculate power
3950 N$(M) = STR$(M):N$(M) = RIGHT$(N$(M),4)
3960 CALL IBWRT (PSUPP%, N$(M)) 'downstep power supply
3970 \ O = 0
3980 FOR K = 1 TO 5
                      'delay 40 ms
         0 = 0 + 1
4000 NEXT K
4010 NEXT M
4020
4030 FOR M = D TO H STEP -15 'measure from 70% to 3% Voc
4040 I = I + 1
4050 \text{ CHAN}% = 0: CTRL% = 0: MODE% = 0
4070 \text{ STAT} = 0
4080 DEF SEG = DSEG
4090 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, V%(0), STAT%)
4100
        VTOT = 0
        FOR X = 0 TO 4
4110
                 VTOT = V%(X) + VTOT 'average 5 samples
4120
4130
        NEXT X
4140 DEF SEG = DATA.SEGMENT
4150 CALL IBRD(DMM%, Z$)
                          'calculate current
4160 \text{ CURR}(I) = VAL(Z\$)
4170 VOLT(I) = VTOT/2048 - Cl 'calculate voltage
4180 PMAX(I) = VOLT(I) * CURR(I) 'calculate power
4190 \text{ NS}(M) = \text{STR}(M):NS(M) = \text{RIGHT}(NS(M), 4)
4200 CALL IBWRT (PSUPP%, N$(M)) 'downstep power supply
4210 0 = 0
4220 \text{ FOR } K = 1 \text{ TO } 5
          0 = 0 + 1 'delay 40 ms
4230
4240 NEXT K
4250 NEXT M
4260
4270 E = M
4280 FOR M = E TO 1000 STEP -2 'measure from 3% Voc to 0
4290 I = I + 1
4300 \text{ CHAN}% = 0: CTRL% = 0: MODE% = 0
4310 STOR% = 0: COUNT = 5: RATE = 500 'sample @ 500 hz
4320 \text{ STAT} = 0
4330 DEF SEG = DSEG
```

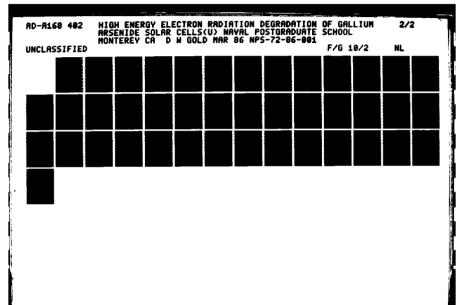
```
4340 CALL AINM(ADAPT%, DEV%, CHAN%, CTRL%, MODE%, STOR%, COUNT,
     RATE, V% (0), STAT%)
        V = TOTV
4350
4360
        FOR X = 0 TO 4
                 VTOT = V%(X) + VTOT 'average 5 samples
4370
        NEXT X
4380
4390 DEF SEG = DATA.SEGMENT
4400 CALL IBRD(DMM%,Z$)
4410 \text{ CURR}(I) = VAL(Z\$)
                           'calculate current
4420 VOLT(I) = VTOT/2048 - Cl 'calculate voltage
4430 \text{ PMAX}(I) = \text{VOLT}(I) * \text{CURR}(I)
                                    'calculate power
4440 \text{ N}(M) = STR(M):N(M) = RIGHT(N(M), 4)
4450 CALL IBWRT (PSUPP%, N$(M)) 'downstep power supply
                                  'check for negative voltage
4460 IF VOLT(I) < 0 THEN 4490
4470 NEXT M 'if not negative, step 3 mV lower
4480
4490 R = I
4500 B = M
4510 FOR M = B TO (J-20) STEP 20 'drive power supply back
4520 N$(M) = STR$(M): N$(M) = RIGHT$(N$(\bar{M}),4) 'to Voc
4530 DEF SEG = DATA.SEGMENT 'output
4540 CALL IBWRT (PSUPP%, N$(M))
4550 NEXT M
4560 \text{ FOR M} = (J-20) \text{ TO J STEP } 1
4570 \text{ N}(M) = STR(M):N(M) = RIGHT(N(M), 4)
4580 DEF SEG = DATA.SEGMENT
4590 CALL IBWRT (PSUPP%, N$(M))
4600 NEXT M
4610
          ******END PARAMETER MEASURING LOOP********
4620 1
4630
          ******DISCONNECT CELL FROM TEST CIRCUIT*******
4640
4650
4660 BEEP
4670 PRINT "TESTING COMPLETED. MOVE SWITCH FROM `TEST'
      TO `SETUP POSITION"
4680 PRINT "TO DISCONNECT CELL FIXTURE FROM TEST CIRCUIT.
     PRESS ENTER KEY"
4690 INPUT "WHEN DONE >",C$
4700 DEF SEG = DATA.SEGMENT
4710 B$=STR$(1314) : A$ = RIGHT$(B$,4) 'drive power supply
4720 CALL IBWRT(PSUPP%,A$) 'to zero volts output
4730 \text{ PMAX} = 0
4740 \text{ FOR T} = 1 \text{ TO } (J-D)
          IF PMAX >= PMAX(T) THEN 4780
          PMAX = PMAX(T) 'find max power point
4760
          VMAX = VOLT(T):IMAX = CURR(T) 'find Vmax,Imax
4770
4780 NEXT T
4790 \text{ ISC} = \text{CURR}(R-1)
 4800 \text{ FF} = PMAX/(VOC * ISC)
 4810 EFF = (PMAX / (.1353 * L * W)) * 100
```

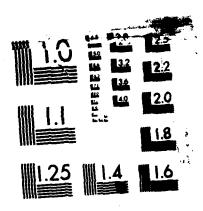
```
4820 ISC = (INT(ISC * 1000))/1000
4830 VOC = (INT(VOC * 1000))/1000
4840 \text{ PMAX} = (INT(PMAX * 1000))/1000
4850 \text{ VMAX} = (INT(VMAX * 1000))/1000
4860 \text{ IMAX} = (INT(IMAX * 1000))/1000
4870 \text{ FF} = (INT(FF * 1000))/1000
4880 \text{ EFF} = (INT(EFF * 10))/10
4890 RETURN
4900
4910
          ******DATA PRINT SUBROUTINE***********
4920
4930 CLS
4940 \text{ FOR I} = 1 \text{ TO R-1}
4950 PRINT USING "##.####"; VOLT(I), CURR(I), PMAX(I)
4960 NEXT I
4970 PRINT
4980 PRINT
4990 BEEP
5000 INPUT "PRESS ENTER TO RETURN TO MENU >",C$
5010 RETURN
5020
5030
          ******SCREEN GRAPHING SUBROUTINE***********
5040
5050 SCREEN 2
5060 CLS
5070 VIEW (100,40 ) - (550,180), ,1 'establish screen limit
5080 WINDOW (0,0) - (VOC + .1, ISC + .01) 'define scale
5090 \text{ FOR I} = 1 \text{ TO R-1}
5100
         PSET (VOLT(I), CURR(I)) 'plot I-V curve
5110 NEXT I
5120 PRINT
5130 BEEP
5140 INPUT "PRESS ENTER TO RETURN TO MENU >",C$
5150 RETURN
5160
5170
          ********P PLOTTING SUBROUTINE*************
5180
5190 BEEP
5200 CLS
5210 INPUT "PRESS ENTER KEY WHEN PLOTTER IS READY >",C$
5220 CLS
5230 OPEN "COM2:9600,S,7,1,RS,CS65535,DS,CD" AS #1
5240 PRINT #1,"IN;SP1;IP2000,2400,8400,7000;"
5250 PRINT #1, "SC0, 1200, 3, 150;" 'initialization
5260 BEEP
5270 PRINT "DO YOU WANT OUTLINE, AXES, AND
                                               HEADER"
5280 INPUT "INCLUDED ON THE GRAPH (Y/N) >", W$
5290 PRINT
5300 IF (W$ = "Y") OR (W$ = "y") THEN 5320
5310 GOTO 5490
5320 PRINT #1, "PUO, OPD1200, 0, 1200, 150, 0, 150, 0, 0PU"
```

```
5330 PRINT #1, "SI.2,.3; TL1.5,0" 'print tick marks
5340 \text{ FOR } X = 0 \text{ TO } 1200 \text{ STEP } 100
5350 PRINT #1, "PA"; X, ", 0; XT; "
5360 IF X<100 THEN PRINT #1, "CP-1.3,-1; LB"; X; CHR$(3)
5370 IF X<1000 AND X>99 THEN PRINT #1, "CP-2.3,-1; LB"; X;
      CHR$(3)
5380 IF X>999 THEN PRINT #1, "CP-2.8, -1; LB"; X; CHR$(3)
5390 NEXT X
5400 \text{ FOR } Y = 0 \text{ TO } 150 \text{ STEP } 15
5410 PRINT #1, "PA 0,", Y, "YT;"
5420 IF Y<100 THEN PRINT #1, "CP-3, -. 25; LB"; Y; CHR$(3)
5430 IF Y>99 THEN PRINT #1, "CP-4, -. 25; LB"; Y; CHR$(3)
5440 NEXT Y
5450 PRINT #1, "SI.35,.5"
5460 PRINT #1, "PA400,0; CP-1,-1.8; LBVOLTAGE (mV)"+CHR$(3)
5470 PRINT #1, "PAO, 45; DIO, 1; CP-1, 1.4; LBCURRENT (mA)"
      +CHR$(3)
5480 PRINT #1, "DI; PU"
5490 \text{ FOR I} = 1 \text{ TO R-1}
5500 PRINT #1, "PA"; INT(VOLT(I)*1000); INT(CURR(I)*1000); "PD"
5510 NEXT I
5520 PRINT #1, "PU"
5530 BEEP
5540 INPUT "DO YOU WANT PARAMETERS INCLUDED (Y/N) >",W$
5550 PRINT
5560 IF (W$ = "Y") OR (W$ = "y") THEN 5580
5570 GOTO 5720
5580 PRINT #1, "SP2; SI.15,.25"
5590 PRINT #1, "PU; PA204, 75; CPO, 0; LBISC = "; ISC*1000; CHR$(3)
5600 PRINT #1, "CP; LBVoc = "; VOC*1000; CHR$(3)
5610 PRINT #1, "CP; LBPmax = "; PMAX*1000; CHR$(3)
5620 PRINT #1, "CP; LBVmp = "; VMAX*1000; CHR$(3)
5630 PRINT #1, "CP; LBImp = "; IMAX*1000; CHR$(3)
5640 PRINT #1, "CP; LBF.F. = "; FF; CHR$(3)
5650 PRINT #1, "CP; LBEFF = "; EFF; CHR$(3)
5660 PRINT #1,"PU;PA 504, 75;CPO,0;LBmA";CHR$(3)
5670 PRINT #1, "CP; LBmV"+CHR$(3)
5680 PRINT #1, "CP; LBmW"+CHR$(3)
5690 PRINT #1, "CP; LBmV"+CHR$(3)
5700 PRINT #1, "CP; LBmA"+CHR$(3)
5710 PRINT #1, "CP; CP; LB%"+CHR$(3)
5720 PRINT #1, "SP2; SI.15,.25"
5730 BEEP
5740 INPUT "DO YOU WANT CELL ID AND DATE INCLUDED (Y/N) >",W$
5750 PRINT
5760 IF (W$ = "Y") OR (W$ = "y") THEN 5780
5770 GOTO 5950
5780 PRINT #1, "PU; PA 828, 140; CPO, 0; LBCELL ID: "; CELLS;
      CHR$(3)
5790 PRINT #1, "CP; LBDATE : "; DATE$; CHR$(3)
5800 BEEP
```

```
5810 INPUT "DO YOU WANT IRRADIATION STATEMENT INCLUDED (Y/N) >",W$
5820 IF (W$ = "Y") OR (W$ = "y") THEN 5850
5830 GOTO 5950
5840 BEEP
5850 INPUT
            "TYPE 1 FOR BEFORE IRRADIATION OR 2 FOR AFTER
     >",ANS$
5860 IF ANS$ = "1" THEN 5890
5870 TF ANS$ = "2" THEN 5910
5880 GOTO 5950
5890 PRINT #1, "CP; LBBEFORE IRRADIATION"+CHR$(3)
5900 GOTO 5950
5910 PRINT #1, "CP; LBAFTER IRRADIATION BY"+CHR$(3)
5920 BEEP
5930 INPUT "ENTER EXPONENT FOR ELECTRON FLUENCE >",NR$
5940 PRINT #1, "CP; LB10 E"+NR$+ " electrons/cm^2 "+CHR$(3)
5950 PRINT #1, "PAO, 0, ; SPO; "
5960 CLOSE
5970 RETURN
5980
5990 1
         ******DATA TO DISK WRITE SUBROUTINE*******
6000
6010 CLS
6020 BEEP
6030 INPUT "PLACE FLOPPY DISK IN DRIVE A. PRESS
                                                   ENTER
     READY >",C$
6040 FILE$ = "A:" + CELL$ + ".DAT"
6050 OPEN FILE$ FOR OUTPUT AS #2
6060 WRITE #2,L,W
6070 \text{ FOR I} = 1 \text{ TO R}
6080 WRITE #2, VOLT(I), CURR(I), PMAX(I)
6090 NEXT I
6100 WRITE #2,30,30,30 'include end of file flag
6110 CLOSE
6120 RETURN
6130
6140
         ******PARAMETER PRINT SUBROUTINE***********
6150
6160 CLS
6170 PRINT
6180 PRINT
6190 PRINT
6200 PRINT
6210 PRINT "SOLAR CELL PARAMETERS FOR CELL "; CELL$; " ARE: "
6220 PRINT
6230 PRINT "VOC="; VOC; " V", "Isc="; ISC; " A", "PMAX="; PMAX; " W"
6240 PRINT "IMAX="; IMAX;" A", "VMAX="; VMAX;"
     FACTOR=";FF
6250 PRINT "EFFICIENCY="; EFF; " %"
6260 PRINT
6270 PRINT
```

```
6280 BEEP
6290 INPUT "PRESS ENTER TO RETURN TO MENU >",C$
6300 RETURN
6310
6320 ′
         ******FLOPPY DISK READ SUBROUTINE********
6330
6340 CLS
6350 BEEP
6360 PRINT "PLACE FLOPPY DISK IN DRIVE A, AND ENTER THE NAME"
6370 INPUT "OF THE DATA FILE TO BE READ >", CELL$
6380 FILE$ = "A:" + CELL$ + ".DAT"
6390 OPEN FILE$ FOR INPUT AS #2
6400 INPUT #2,L,W
6410 I = 1: PMAX = 0
6420 INPUT #2, VOLT(I), CURR(I), PMAX(I)
6430 IF VOLT(I) > 29 THEN 6490
           IF PMAX \geq PMAX(I) THEN 6470
6440
           PMAX = PMAX(I)
6450
           VMAX = VOLT(I) : IMAX = CURR(I)
6460
6470 I = I + 1
6480 GOTO 6420
6490 CLOSE
6500 R = I
6510 ISC = CURR(R-1)
6520 \text{ VOC} = \text{VOLT}(1)
6530 \text{ FF} = PMAX/(VOC * ISC)
6540 \text{ EFF} = (PMAX / (.1353 * L * W)) * 100
6550 \text{ ISC} = (INT(ISC * 1000))/1000
6560 \text{ VOC} = (INT(VOC * 1000))/1000
6570 \text{ PMAX} = (INT(PMAX * 1000))/1000
6580 \text{ VMAX} = (INT(VMAX * 1000))/1000
6590 IMAX = (INT(IMAX * 1000))/1000
         = (INT(FF * 1000))/1000
6600 FF
6610 \text{ EFF} = (INT(EFF * 10))/10
6620 RETURN
```





MICROCOPY RESOLUTION TESTS CHART NATIONAL BURLAND OF STANDARD FROM

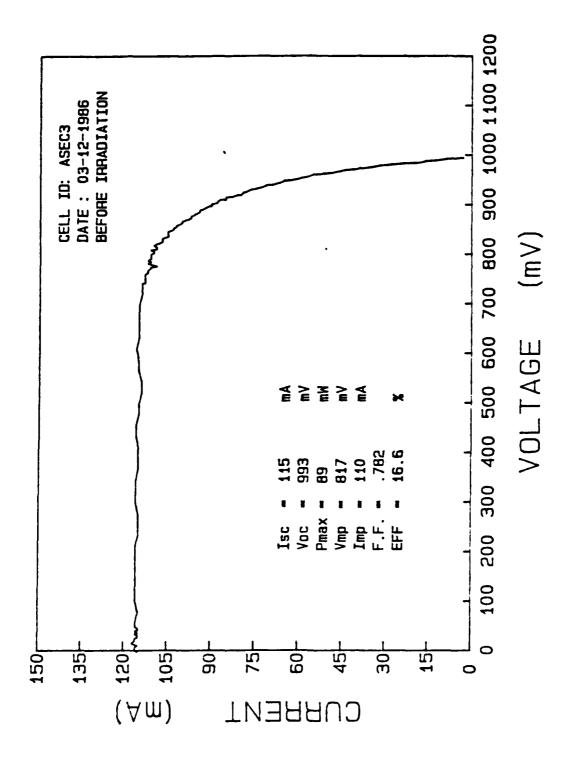


Figure 21. Pre-Irradiation I-V Curve for ASEC Cell Number 3.

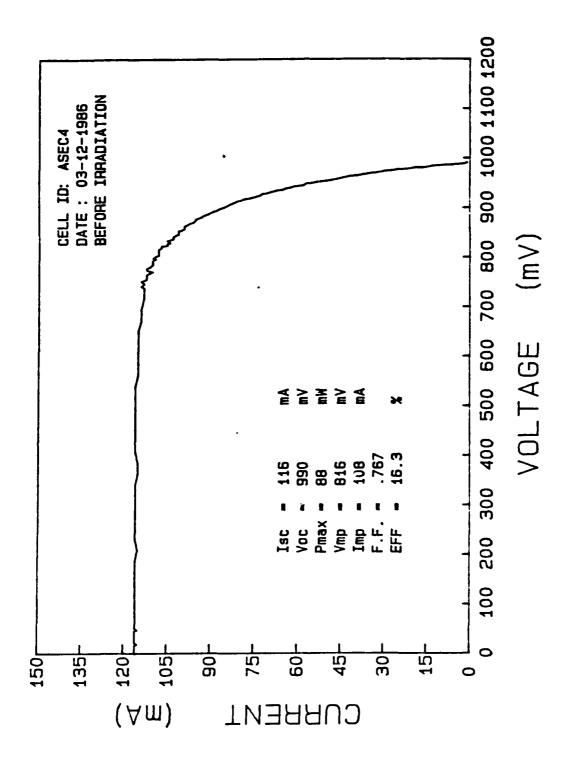


Figure 22. Pre-Irradiation I-V Curve for ASEC Cell Number 4.

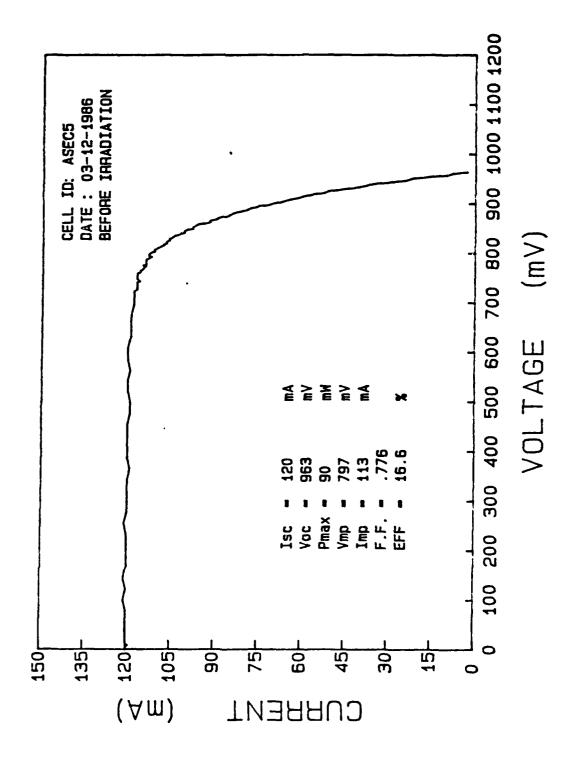


Figure 23. Pre-Irradiation I-V Curve for ASEC Cell Number 5.

Successive processes and action processes

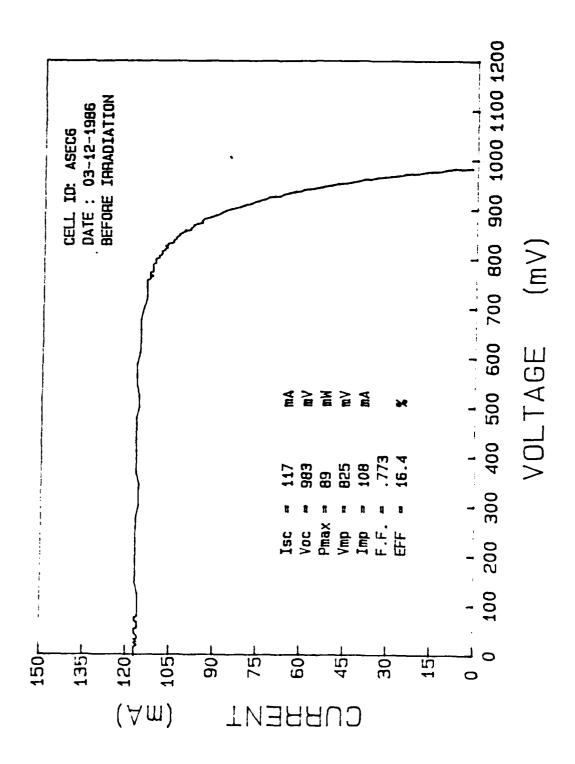


Figure 24. Pre-Irradiation I-V Curve for ASEC Cell Number 6.

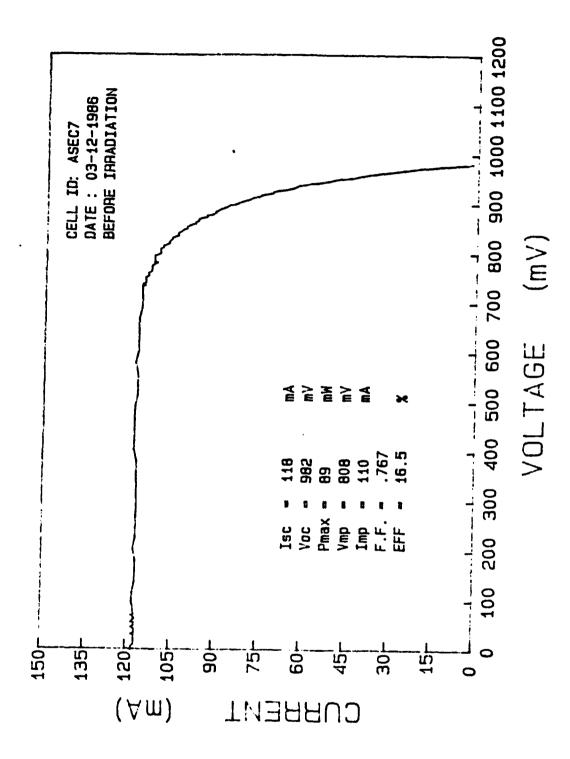


Figure 25. Pre-Irradiation I-V Curve for ASEC Cell Number 7.

reseased besises securities properties assessed becomes and

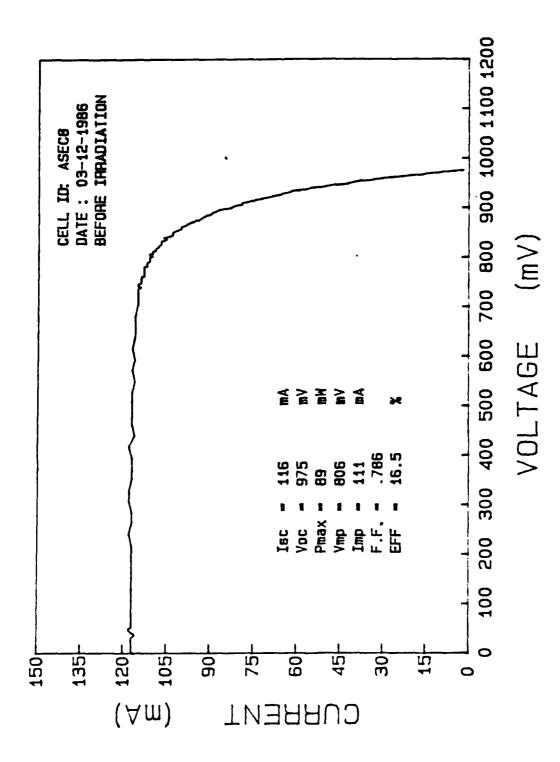


Figure 26. Pre-Irradiation I-V Curve for ASEC Cell Number 8.

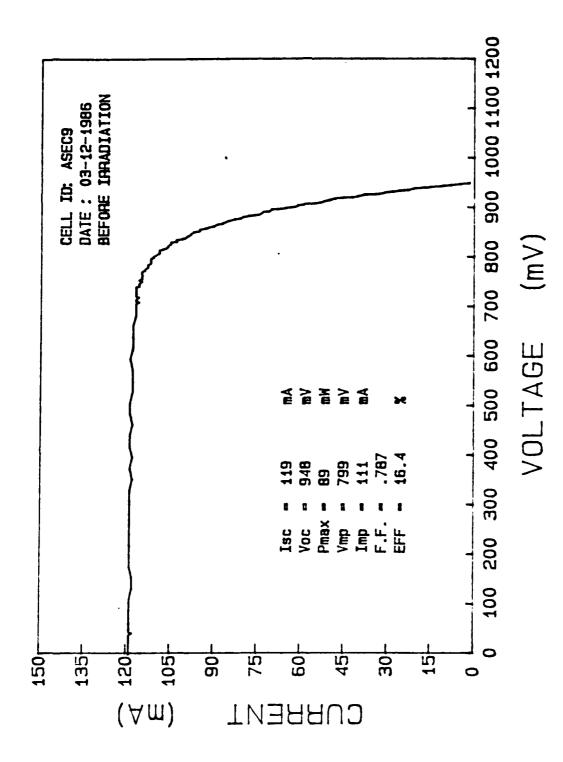


Figure 27. Pre-Irradiation I-V Curve for ASEC Cell Number 9.

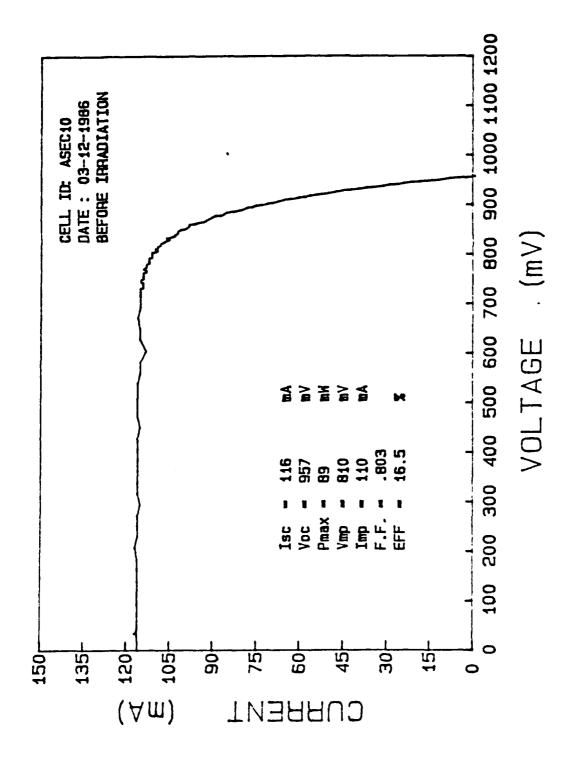


Figure 28. Pre-Irradiation I-V Curve for ASEC Cell Number 10.

APPENDIX F POST-IRRADIATION CELL TEST RESULTS

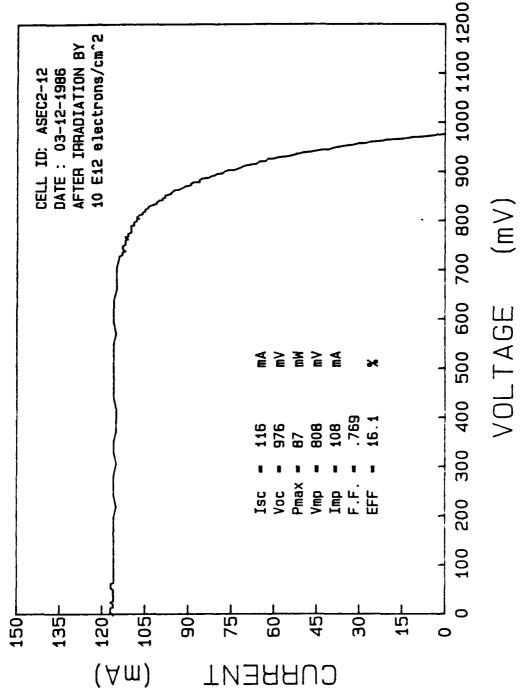


Figure 29. Post-Irradiation I-V Curve for ASEC Cell Number 2 After Irradiation by 1012 e/cm2.

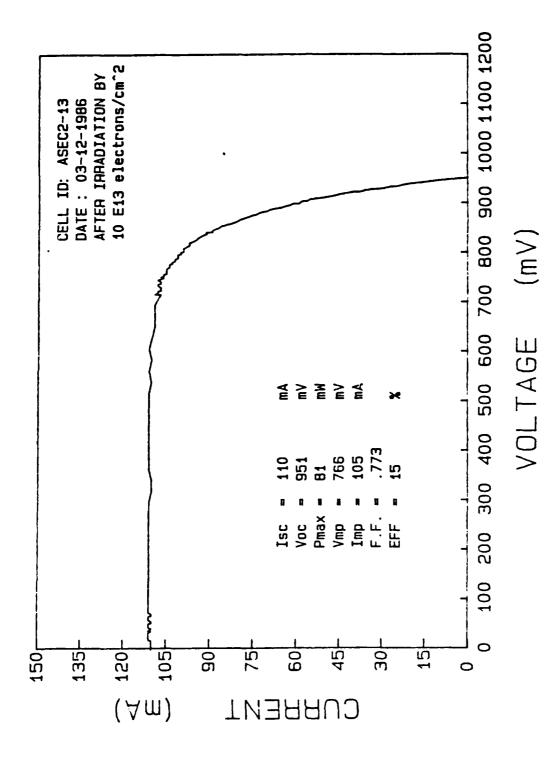
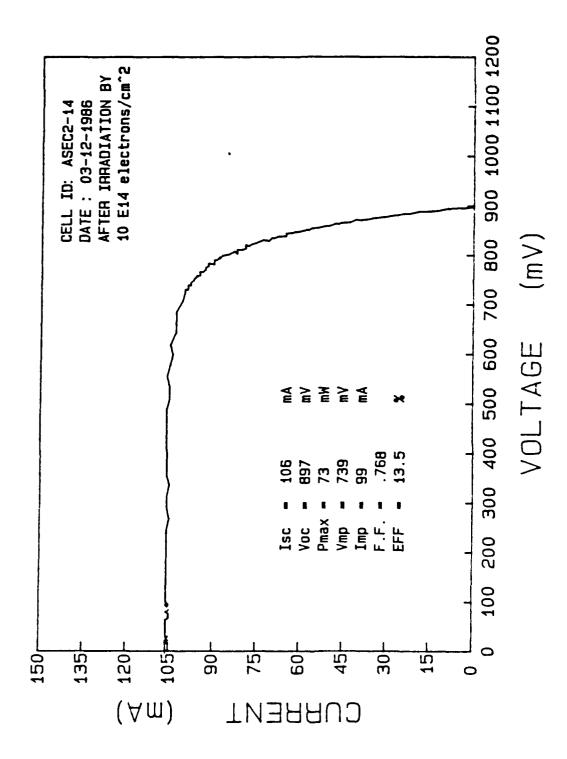


Figure 30. Post-Irradiation I-V Curve for ASEC Cell Number 2 After Irradiation by 10^{13} e/cm².



geres proposes provided respected corrects lighted besides besides respecte executes assessed

Figure 31. Post-Irradiation I-V Curve for ASEC Cell NUmber 2 After Irradiation by 10^{14} e/cm².

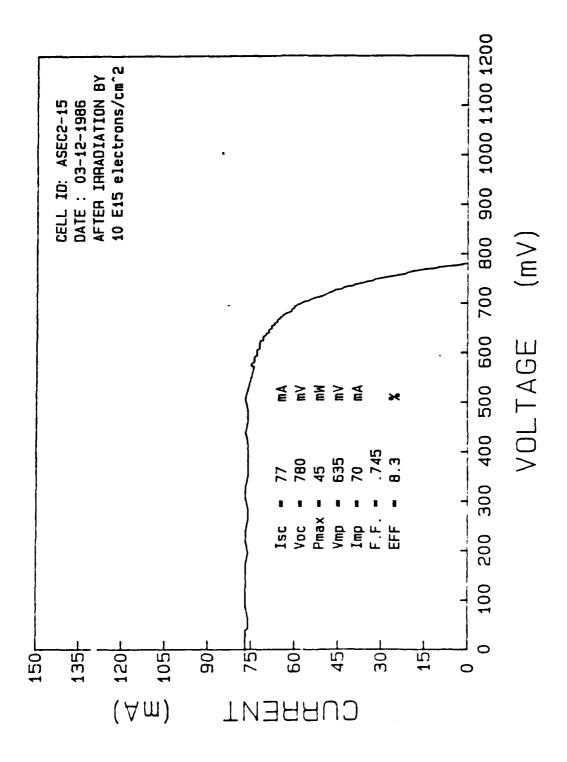
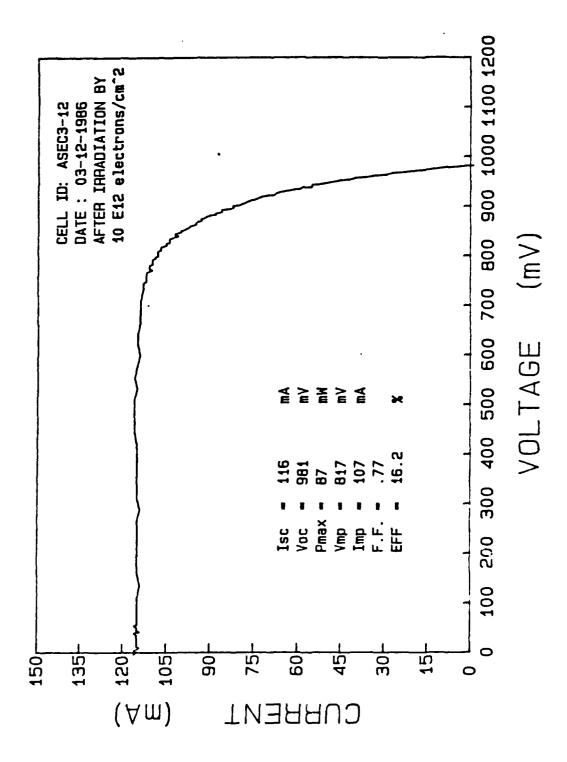


Figure 32. Post-Irradiation I-V Curve for ASEC Cell Number 2 After Irradiation by 1015 e/cm2.



\$5550 E3550555 E555555

Figure 33. Post-Irradiation I-V Curve for ASEC Cell Number 3 After Irradiation by 1012 e/cm².

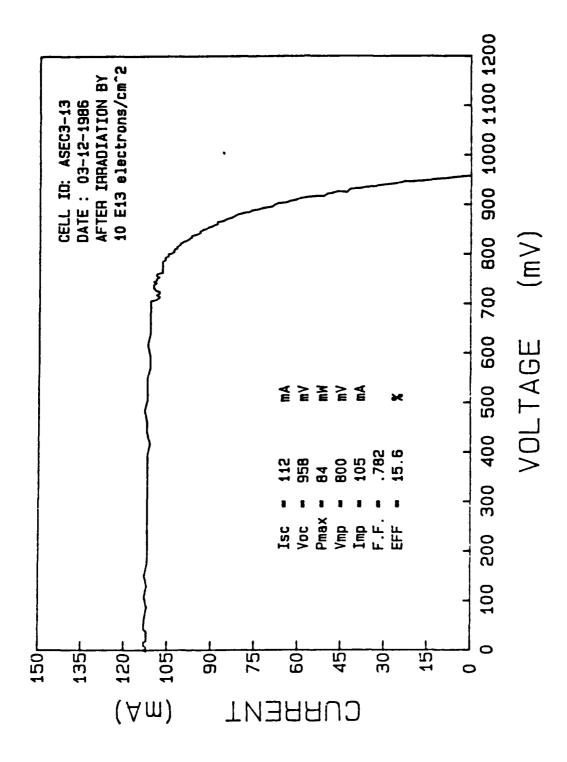
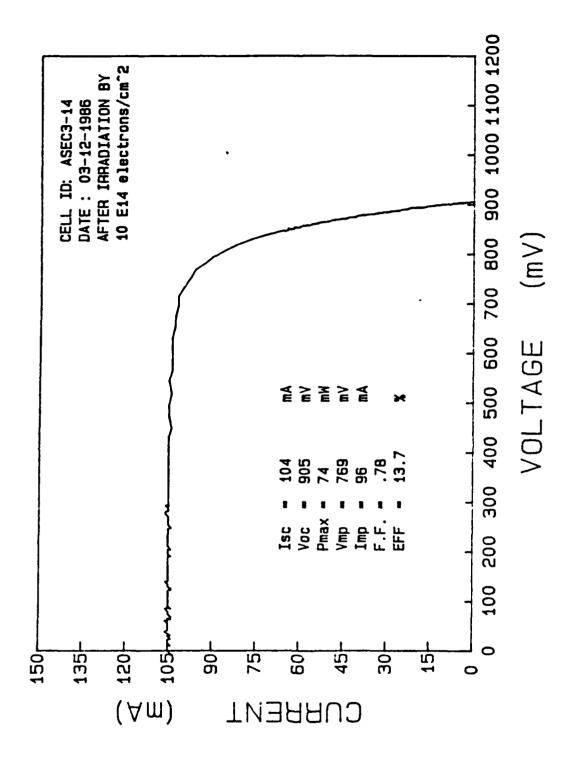


Figure 34. Post-Irradiation I-V Curve for ASEC Cell Number 3 After Irradiation by 10^{13} e/cm².



passed terrorial presentation properties properties properties and properties and

Figure 35. Post-Irradiation I-V Curve for ASEC Cell Number 3 After Irradiation by 1014 e/cm².

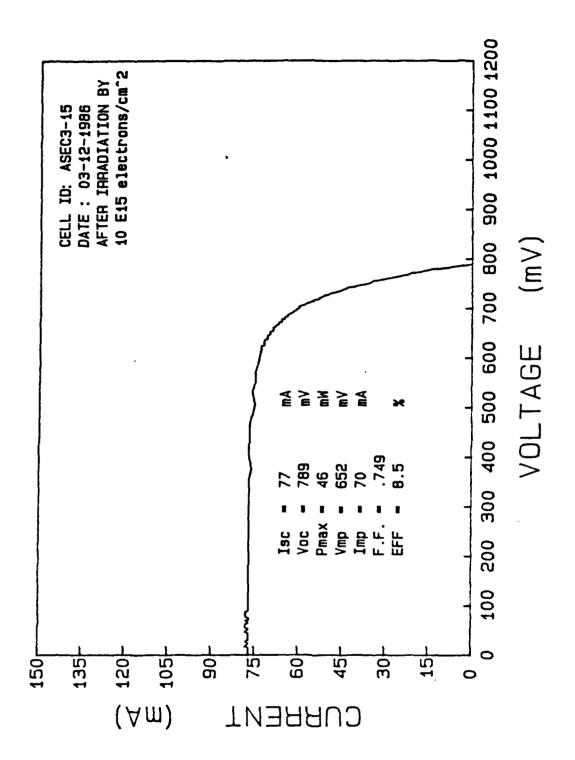


Figure 36. Post-Irradiation I-V Curve for ASEC Cell Number 3 After Irradiation by 1015 e/cm2.

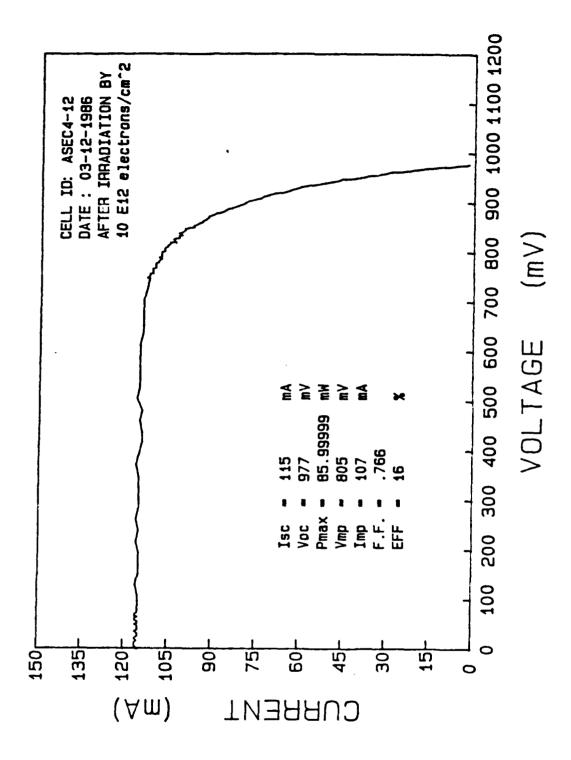


Figure 37. Post-Irradiation I-V Curve for ASEC Cell Number 4 After Irradiation by 10^{12} e/cm^2 .

Paradical regionery processes increases any anymous figures.

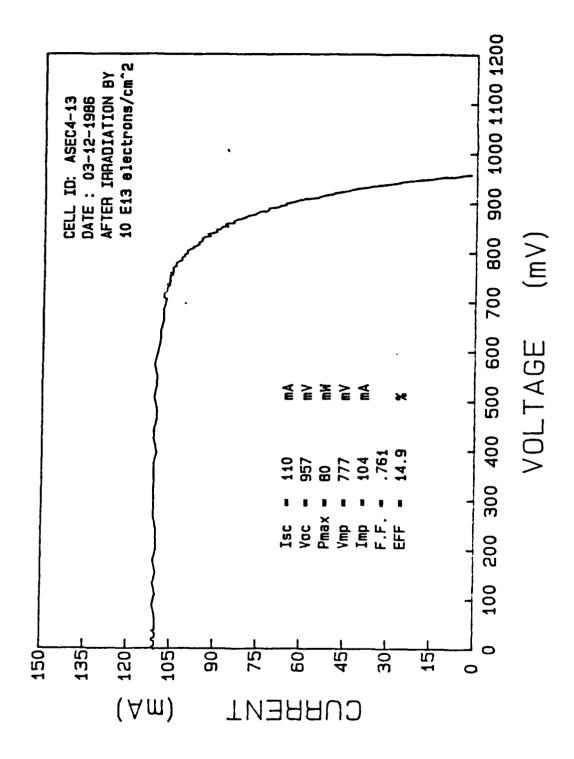


Figure 38. Post-Irradiation I-V Curve for ASEC Cell Number 4 After Irradiation by 10^{13} e/cm².

passi consist consist manager passages sensions reasoned because income income instance has

Figure 39. Post-Irradiation I-V Curve for ASEC Cell Number 4 After Irradiation by 1014 e/cm2.

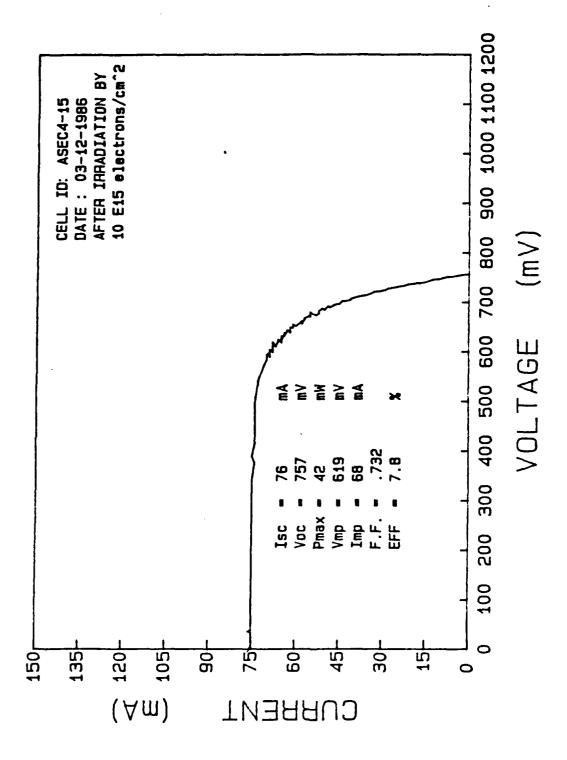


Figure 40. Post-Irradiation I-V Curve for ASEC Cell Number 4 After Irradiation by $10^{15}~\rm e/cm^2$.

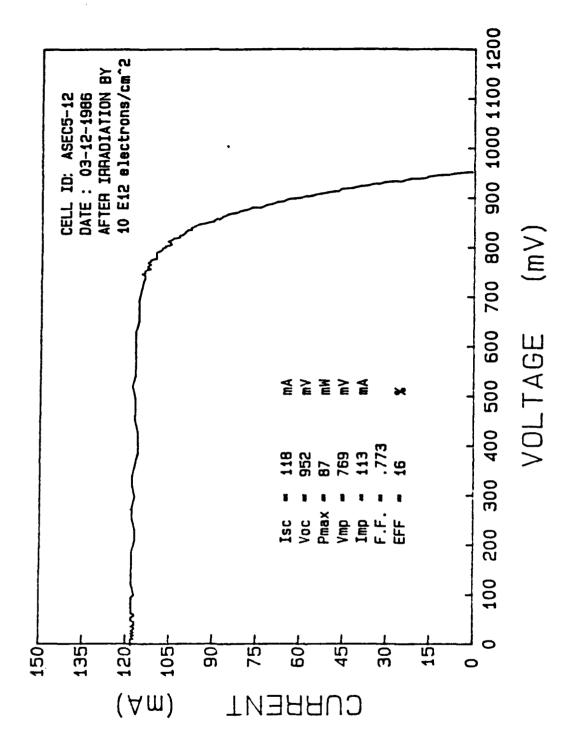


Figure 41. Post-Irradiation I-V Curve for ASEC Cell Number 5 After Irradiation by $10^{12}~\rm e/cm^2$.

Accessed to accessed the property of the second of the second of the second sec

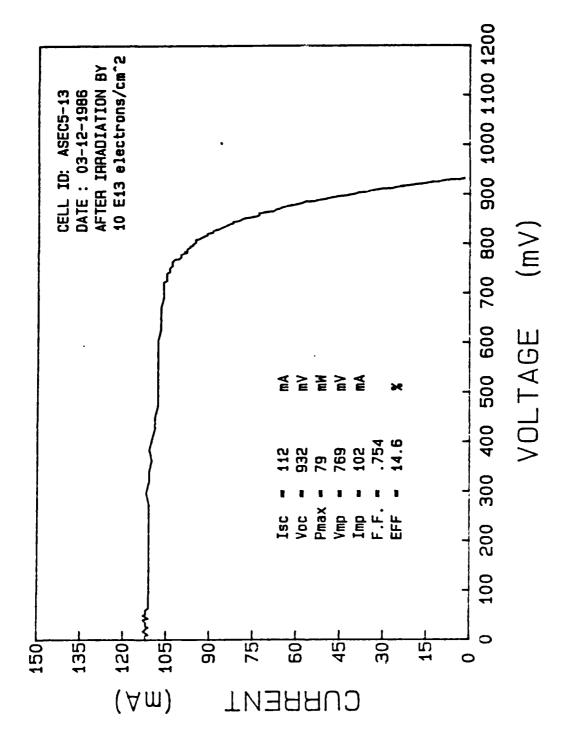


Figure 42. Post-Irradiation I-V Curve for ASEC Cell Number 5 After Irradiation by $10^{13}~\rm e/cm^2$.

Secretaria (

CONTRACTOR OF BEEN SECOND

Figure 43. Post-Irradiation I-V Curve for ASEC Cell Number 5 After Irradiation by 10¹⁴ e/cm².

CONTRACT AND PROPERTY DESCRIPTION OF STREET, S

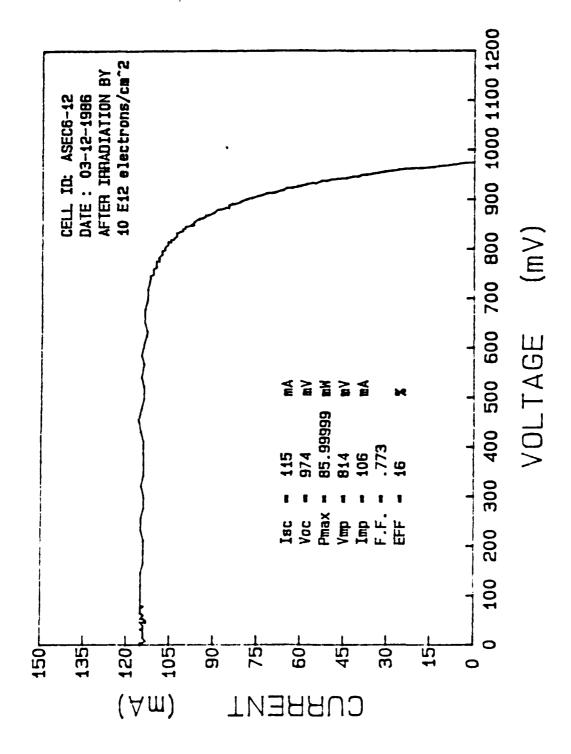
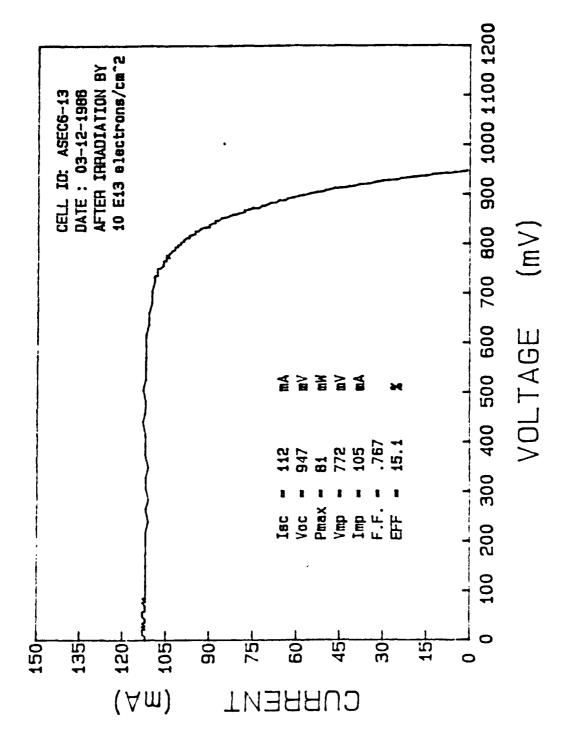


Figure 44. Posc-Irradiation I-V Curve for ASEC Cell Number 6 After Irradiation by 10^{12} e/cm².



prose received measure approve approved assurance

Figure 45. Post-Irradiation I-V Curve for ASEC Cell Number 6 After Irradiation by $10^{13}~\rm e/cm^2$.

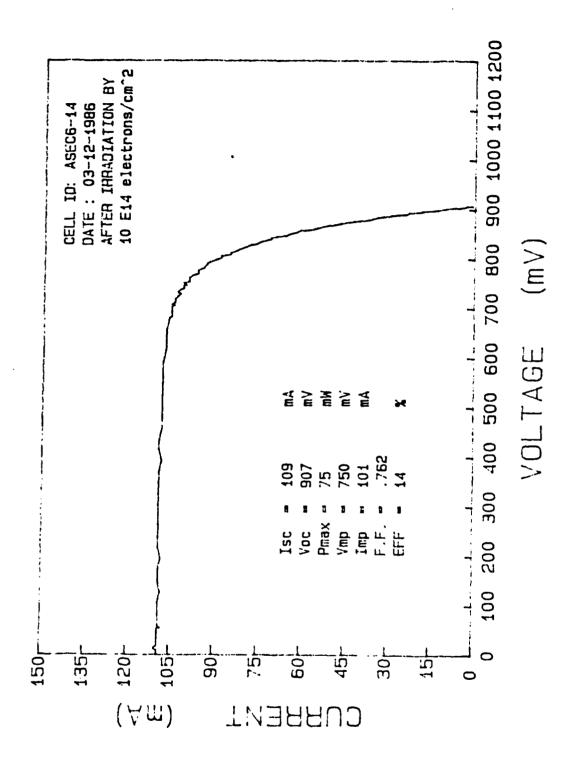


Figure 46. Post-Irradiation I-V Curve for ASEC Cell Number 6 After Irradiation by $10^{14}~\rm e/cm^2$.

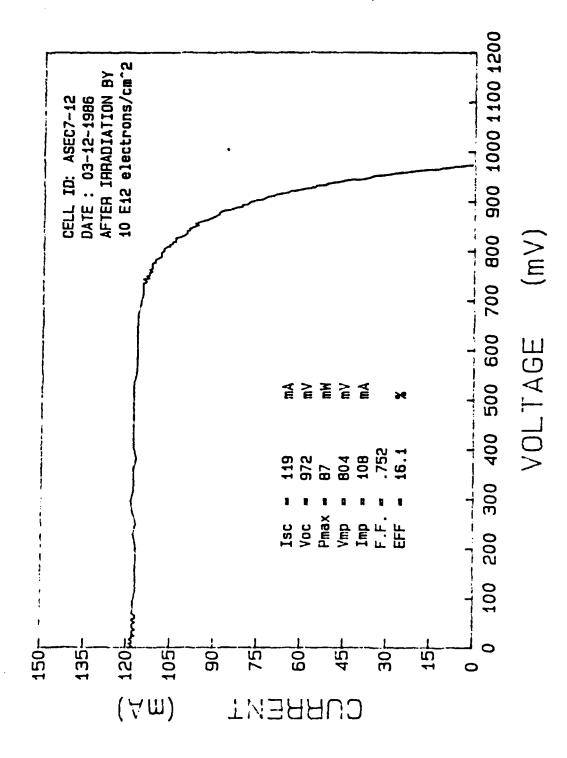


Figure 47. Post-Irradiation I-V Curve for ASEC Cell Number 7 After Irradiation by 10^{12} e/cm².

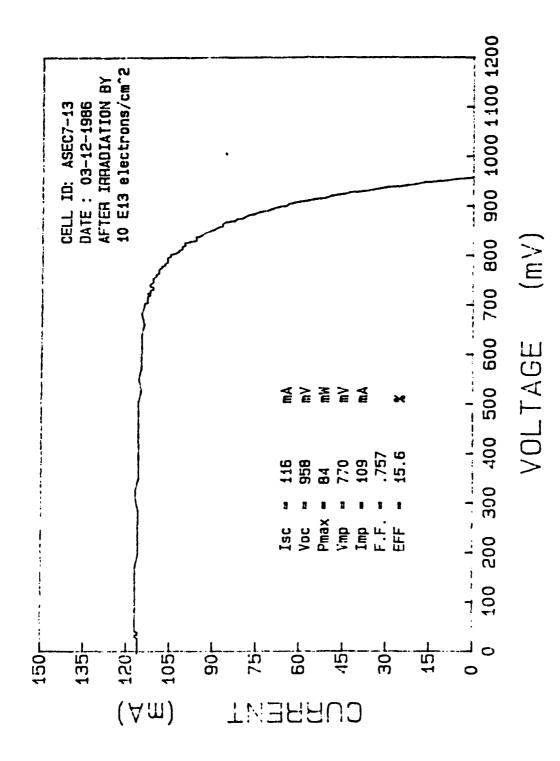


Figure 48. Post-Irradiation I-V Curve for ASEC Cell Number 7 After Irradiation by 1013 e/cm2.

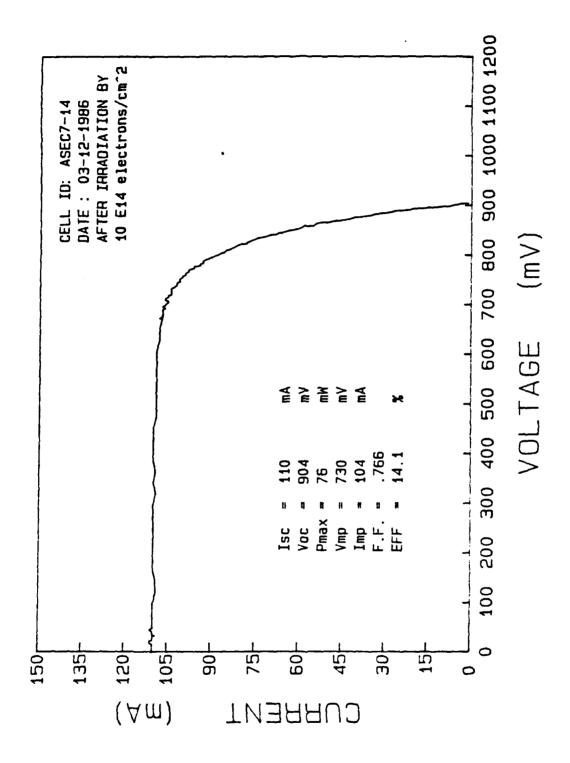


Figure 49. Post-Irradiation I-V Curve for ASEC Cell Number 7 After Irradiation by 1014 e/cm2.

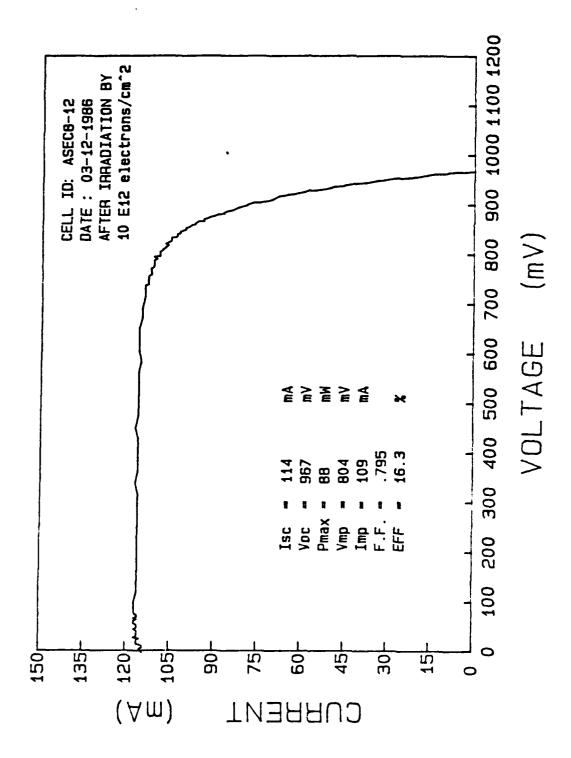


Figure 50. Post-Irradiation I-V Curve for ASEC Cell Number 8 After Irradiation by $10^{12}~\rm e/cm^2$.

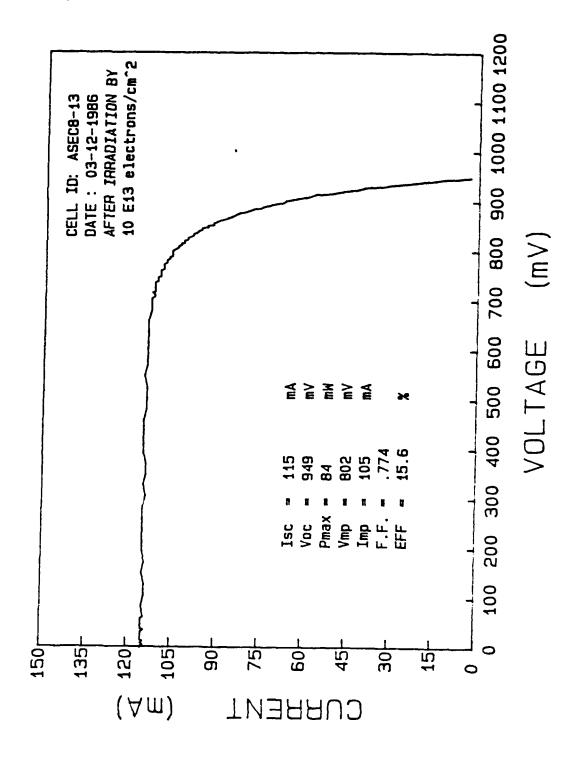


Figure 51. Post-Irradiation I-V Curve for ASEC Cell Number 8 After Irradiation by 1013 e/cm2.

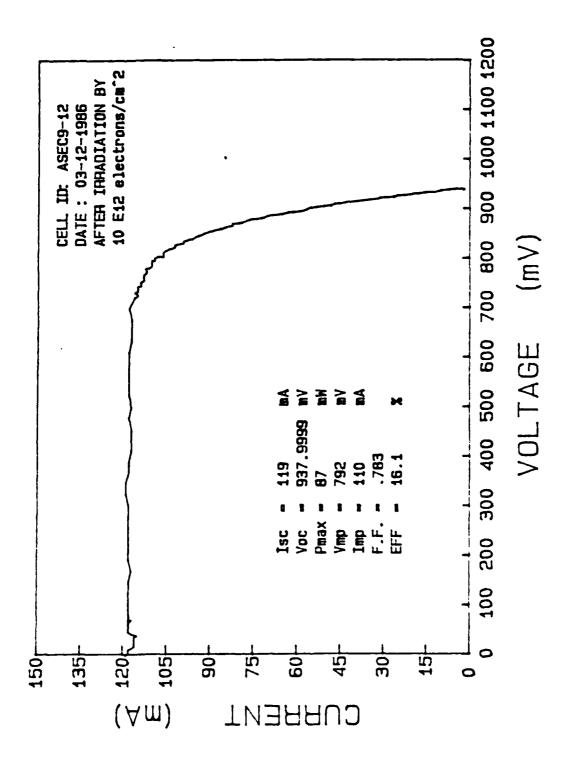
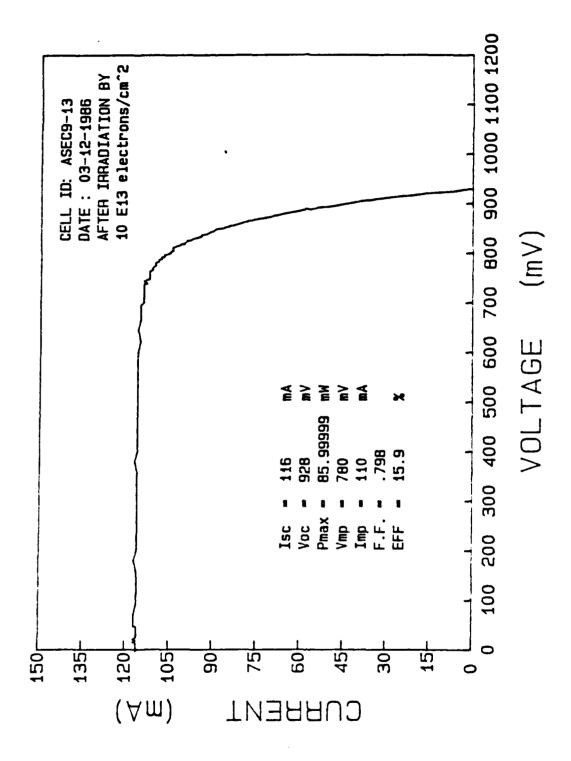


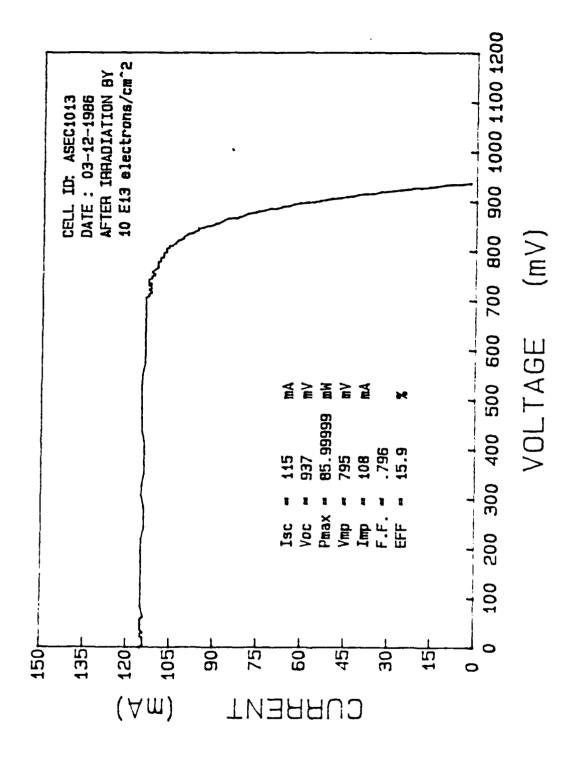
Figure 52. Post-Irradiation I-V Curve for ASEC Cell Number 9 After Irradiation by 10^{12} e/cm².



ADDAD GEOGRAPH HERECHEST HERECHES CONTRACTOR

Figure 53. Post-Irradiation I-V Curve for ASEC Cell Number 9 After Irradiation by 10^{13} e/cm².

Figure 54. Post-Irradiation I-V Curve for ASEC Cell Number 10 After Irradiation by 1012 e/cm2.



especially decesses processes.

Figure 55. Post-Irradiation I-V Curve for ASEC Cell Number 10 After Irradiation by 1013 e/cm2.

LIST OF REFERENCES

- 1. Mabie, K. T., Solar Simulation Laboratory Description and Manual, M.S. Thesis, Naval Postgraduate School, Monterey, California, June 1985.
- 2. Rauschenbach, H.S., · <u>Solar Cell Array Design Handbook</u>, Van Nostrand Reinhold Co., 1980.
- 3. Personal communication with Dr. P. Iles, Applied Solar Energy Corporation, 02 October 1985.
- 4. Applied Solar Energy Corporation letter to the Naval Postgraduate School, Code 67Fu, Subject: Details of Gallium Arsenide Solar Cells, 14 January 1986.
- 5. Tada, H.Y. and others, <u>Solar Cell Radiation Handbook</u>
 Jet Propulsion Laboratory Publication 82-69, 1 November
 1982.
- 6. Anspaugh, B. E. and R. G. Downing, Radiation Effects in Silicon and Gallium Arsenide Solar Cells Using Isotropic and Normally Incident Propulsion Laboratory Publication 84-61, 1 September 1984.
- 7. Kratos Analytical Instruments, <u>UV-VIS-IR Monochromators</u> and Illumination Systems, 1985.
- 8. Hewlett Packard Company, Model 3478A Digital Multimeter Service Manual, 1981.
- 9. Hewlett Packard Company, Model 6825A Bipolar Power Supply/Amplifier, Operating and Service Manual, 1974.
- 10. Hewlett Packard Company, Model 59501B HP-IB Isolated DAC/Power Supply Programmer, Operating and Service Manual, 1983.
- 11. International Business Machines Corporation, <u>PC Options</u> and <u>Adapters Technical Reference Manual Data Acquisition Adapter</u>, 1984.
- 12. National Instruments Corporation, GPIB-PC User Manual For the IBM Personal Computer and Compatibles, October 1984.

13. International Business Machines Corporation, IBM Personal Computer Data Acquisition and Control Programming Support, November 1984.

Secret Assessed Places Assessed Secretary Secretary

property property appropriate property

BIBLIOGRAPHY

- Cannon, T.W. and C. J. Riordan, <u>Photovoltaic Device Performance Calculations From Measured Indoor and Outdoor Spectra</u>, in Seventeenth IEEE Photovoltaic Specialists Conference, 1984.
- Cox, C.H., III, and T. H. Warner, <u>Swept Measurement of High Power I-V Curves</u>, in Sixteenth IEEE Photovoltaic Specialists Conference, 1982
- Cox, C.H., III, and T. H. Warner, <u>Photovoltaic I-V Curve</u> <u>Measurement Techniques</u>, U. S. Department of Energy Report DOE/ET/20279-215, August 1982.
- Goodelle, G.S. and G. R. Brooks, <u>Simulator Spectral Characterization Using Balloon Calibrated Solar Cells With Narrow Band Pass Filters</u>, in Fifteenth IEEE Photovoltaic Specialists Conference, 1981.
- Sacco, S.B., <u>Description</u> of the <u>MIT/Lincoln Laboratory</u> Photovoltaic <u>Systems Test</u> Facility, U.S. Department of Energy Report C00-4094-41, 30 June 1979.
- Soper, R.L., <u>Radiation</u> <u>Shielding for the Naval Postgraduate</u> <u>School LINAC</u>, M. S. Thesis, Naval Postgraduate School, Monterey, CA, June 1967.
- Stefanakos, E.K. and W. J. Collis, <u>Solar Cell Testing and Evaluation</u>, Sandia National Laboratories Report SAND 81-7194, April 1982.

INITIAL DISTRIBUTION LIST

		No.	Copies
1.	Defense Technical Information Center Cameron Station Alexandria, VA 22304-6145		2
2.	Library, Code 0142 Naval Postgraduate School Monterey, CA 93943-5002		2
3.	Distinguished Professor Allen E. Fuhs Code 72 Naval Postgraduate School Monterey, CA 93943-5000		4
4.	Department of the Navy Commander Space and Naval Warfare Systems Command, PDW 106-483 Attn: MAJ. M.L. McNulty Washington, D. C. 20363-5100		5
5.	Dr. Harriett B. Rigas Code 62 Naval Postgraduate School Monterey, CA 93943-5000		2
6.	Naval Space Command, Code N13 Dahlgren, VA 22448		2
7.	Dr. Fred Raymond, Code 9110 Naval Research Laboratory 4555 Overlook Ave., SW Washington, D. C. 20375		1
8.	Dr. Peter A. Iles Chief Scientist, Research and Development Applied Solar Energy Corporation 15251 E. Don Julian Rd. P.O. Box 1212 City of Industry, CA 91749		1

9.	Dr. Bruce Anspaugh Jet Propulsion Laboratory California Institute of Technology 4800 Oak Grove Ave. Pasadena, CA 91109	1
10.	Richard L. Statler, Code 6623 Naval Research Laboratory 4555 Overlook Ave., SW Washington, D. C. 20375	1
11.	Dr. Hans Rauschenbach TRW Defense and Space Systems One Space Park, Ml 1406 Redondo Beach, CA 90278	1
12.	Terry M. Trumble Energy Conversion Branch Aerospace Power Division Aero Propulsion Laboratory Wright-Patterson Air Force Base Ohio, 45433	1
13.	Director of Research Administration Code 012 Naval Postgraduate School Monterey, CA 93943-5000	1
14.	Dr. F. Buskirk Code 61Bs Naval Postgraduate School Monterey, CA 93943-5000	1
15.	United States Space Command Attn: Technical Library Peterson AFB, CO 80914	2
	Professor S. Michael Code 62Mi Naval Postgraduate School Monterey, CA 93943-5000	1
17.	Department of the Navy Commander Space and Naval Warfare Systems Command, PDW 106-5 Attn: LT. D.W. GOLD Washington, D.C. 20363-5100	2